# Ultimate Acceleration in Quantum Mechanics to Explain and Calculate Earth's Geomagnetic Field: 0.6Gauss 

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#### Abstract

In analogy with the ultimate speed $c$, there is an ultimate acceleration $\beta$, nobody's acceleration can exceed this limit $\beta$, in the solar system, $\beta=2.961520 \mathrm{e}+10\left(\mathrm{~m} / \mathrm{s}^{2}\right)$. Because this ultimate acceleration is large, any effect related to $\beta$ will become easy to test, including quantum gravity tests. In this paper, an approach is put forward to connect the ultimate acceleration with quantum theory, and is applied to sunspot, earth's tropic cyclones, and earth's geomagnetic field problems. The sunspot cycle is calculated to be 10.93 years due to the ultimate acceleration. A simulation was carried out, clearly showing the inner structure of a cyclone, which is consistent with the famous DIANA cyclone on 12 September 1984 in situ observation measured by an aircraft. The sunspot structure is also investigated, which has a similarity to that of a tropic cyclone. The ultimate acceleration provides a mechanism to explain geomagnetic field, the earth's geomagnetic field is calculated to be 0.6 Gauss at the north pole.


## 1. Introduction

In general, some quantum gravity proposals [1,2] are extremely hard to test in practice, as quantum gravitational effects are appreciable only at the Planck scale [3]. But ultimate acceleration provides another scheme to deal with quantum gravity effects.

In analogy with the ultimate speed $c$, there is an ultimate acceleration $\beta$, nobody's acceleration can exceed this limit $\beta$, in the solar system, $\beta=2.961520 \mathrm{e}+10\left(\mathrm{~m} / \mathrm{s}^{2}\right)$. Because this ultimate acceleration is large, any effect related to $\beta$ will become easy to test, including quantum gravity tests. In this paper, an approach is put forward to connect the ultimate acceleration with quantum theory, and is applied to sunspot, earth's tropic cyclones, and earth's geomagnetic field problems.

## 2. How to connect the ultimate acceleration with quantum theory

In relativity, the speed of light $c$ is the ultimate speed, nobody's speed can exceed this limit $c$. The relativistic velocity $u$ of a particle in the coordinate system $\left(x_{1}, x_{2}, x_{3}, x_{4}=i c t\right)$ satisfies

$$
\begin{equation*}
u_{1}^{2}+u_{2}^{2}+u_{3}^{2}+u_{4}^{2}=-c^{2} . \tag{1}
\end{equation*}
$$

No matter what particles (electrons, molecules, neutrons, quarks), their 4-vector velocities all have the same magnitude: $|u|=i c$. All particles gain equality because of the same magnitude of the 4 -velocity $u$. The acceleration $a$ of a particle is given by

$$
\begin{equation*}
a_{1}^{2}+a_{2}^{2}+a_{3}^{2}=a^{2} ; \quad\left(a_{4}=0 ; \quad \because x_{4}=i c t\right) \tag{2}
\end{equation*}
$$

Assume that particles have an ultimate acceleration $\beta$ as the limit, no particle can exceed this acceleration limit $\beta$. Subtracting both sides of the above equation by $\beta^{2}$, we have

$$
\begin{equation*}
a_{1}^{2}+a_{2}^{2}+a_{3}^{2}-\beta^{2}=a^{2}-\beta^{2} ; \quad a_{4}=0 \tag{3}
\end{equation*}
$$

It can be rewritten as

$$
\begin{equation*}
\left[a_{1}^{2}+a_{2}^{2}+a_{3}^{2}+0+(i \beta)^{2}\right] \frac{1}{1-a^{2} / \beta^{2}}=-\beta^{2} \tag{4}
\end{equation*}
$$

Now, the particle has an acceleration whose five components are specified by

$$
\begin{array}{ll}
\alpha_{1}=\frac{a_{1}}{\sqrt{1-a^{2} / \beta^{2}}} ; \quad \alpha_{2}=\frac{a_{2}}{\sqrt{1-a^{2} / \beta^{2}}}  \tag{5}\\
\alpha_{3}=\frac{a_{3}}{\sqrt{1-a^{2} / \beta^{2}}} ; \quad \alpha_{4}=0 ; \quad \alpha_{5}=\frac{i \beta}{\sqrt{1-a^{2} / \beta^{2}}}
\end{array}
$$

where $\alpha_{5}$ is the newly defined acceleration in five-dimensional space-time ( $x_{1}, x_{2}, x_{3}, x_{4}=i c t, x_{5}$ ). Thus, we have

$$
\begin{equation*}
\alpha_{1}^{2}+\alpha_{2}^{2}+\alpha_{3}^{2}+\alpha_{4}^{2}+\alpha_{5}^{2}=-\beta^{2} ; \quad \alpha_{4}=0 \tag{6}
\end{equation*}
$$

It means that the magnitude of the newly defined acceleration $\alpha$ for every particle takes the same value: $|\alpha|=i \beta$ (constant imaginary number), all particle accelerations gain equality for the sake of the same magnitude.

How do resolve the velocity $u$ and acceleration $\alpha$ into $x, y$, and $z$ components? In a realistic world, a hand can rotate a ball moving around a circular path at constant speed $v$ with constant centripetal acceleration $a$, as shown in Fig.1(a). Likewise, the $u$ and $\alpha$ let the particle move spirally, as shown in Fig.1(b), projecting out the real $x, y$, and $z$ components.


Fig. 1 (a) A hand rotates a ball moving around a circular path at constant speed $v$ with constant centripetal acceleration $a$. (b) The particle moves along the $x_{1}$ axis with the constant speed $|u|=i c$ in $u$ direction and constant centripetal force in the $x_{5}$ axis at the radius $i R$ (imaginary number).
$<$ Clet2020 Script>//[26].
double $D[100], S[2000] ;$ int $\mathrm{i}, \mathrm{j}, \mathrm{R}, \mathrm{X}, \mathrm{N}$;
int main( $)\{\mathrm{R}=50 ; \mathrm{X}=50 ; \mathrm{N}=600 ; \mathrm{D}[0]=-50 ; \mathrm{D}[1]=0 ; \mathrm{D}[2]=0 ; \mathrm{D}[3]=\mathrm{X} ; \mathrm{D}[4]=0 ; \mathrm{D}[5]=0 ; \mathrm{D}[6]=-50 ; \mathrm{D}[7]=\mathrm{R} ; \mathrm{D}[8]=0$;
$\mathrm{D}[9]=600 ; \mathrm{D}[10]=10 ; \mathrm{D}[11]=\mathrm{R} ; \mathrm{D}[12]=0 ; \mathrm{D}[13]=3645$;
Lattice(SPIRAL,D,S);SetViewAngle( $0,80,-50$ );DrawFrame(FRAME NULL, 1,0xffffff);
Draw("LINE, 0,2, XYZ, $0 ", "-150,0,0,-50,0,0 ") ;$ Draw("ARROW,0,2,XȲZ, $10 ", " 50,0,0,150,0,0 ")$;
SetPen(2,0xff0000);Plot("POLYLINE,0,600,XYZ",S[9]);i=9+3*N-6;Draw("ARROW,0,2,XYZ,10",S[i]);
TextHang(S[i],S[i+1],S[i+2]," \#iflu|=ic\#t");TextHang(150,0,0," \#ifx\#sd1\#t");SetPen(2,0x005fff);
Draw("LINE, 1,2,XYZ, 8","-50,0,50,-50,0,100");Draw("LINE, 1,2,XYZ,8","-40,0,50,-40,0,100");
Draw("ARROW,0,2,XYZ,10","-80,0,100,-50,0,100");Draw("ARROW,0,2,XYZ, $10 ", "-10,0,100,-40,0,100$ ");
TextHang $(-50,0,110, " 1$ spiral step" $) ; i=9+3^{*} N ; S[i]=50 ; S[i+1]=10 ; S[i+2]=10$;

Draw("ARROW,0,2,XYZ,10","50,0,0,50,80,80");TextHang(50,80,80," \#ifx\#sd5\#t");
Draw("ARROW, $0,2, \mathrm{XYZ}, 10$ "," $50,72,0,50,0,72$ "); TextHang(50,0,72," \#ifx\#sd4\#t");
SetPen(3,0x00ffff);Draw("ARROW,0,2,XYZ, $15^{\prime \prime}, S[i-3]$ );TextHang(S[i],S[i+1],S[i+2]," \#if| $\alpha \mid=i \beta \# t "$ );
SetPen(3,0x00ff00);Draw("ARROW,0,2,XYZ, 15","50,0,0,120,0,0");TextHang(110,5,5," \#ifJ\#t");
TextHang(-60,0,-80," right hand chirality"); $\} \# v 07=?>$ A\#t

In analogy with the ball in a circular path, consider a particle in one-dimensional motion along the $x_{I}$ axis at the speed $v$, in Fig.1(b) it moves with the constant speed $|u|=i c$ almost along the $x_{4}$ axis and slightly along the $x_{I}$ axis, and the constant centripetal acceleration $|\alpha|=i \beta$ in the $x_{5}$ axis at the constant radius $i R$ (imaginary number); the coordinate system ( $x_{1}, x_{4}=i c t, x_{5}=i R$ ) establishes a cylinder coordinate system in which this particle moves spirally at the speed $v$ along the $x_{I}$ axis. According to the usual centripetal acceleration formula $a=v^{2} / r$, the acceleration in the $x_{4}-x_{5}$ plane is given by

$$
\begin{equation*}
a=\frac{v^{2}}{r} \Rightarrow i \beta=\frac{|u|^{2}}{i R}=-\frac{c^{2}}{i R}=i \frac{c^{2}}{R} . \tag{7}
\end{equation*}
$$

Therefore, the track of the particle in the cylinder coordinate system $\left(x_{1}, x_{4}=i c t, x_{5}=i R\right)$ forms a shape, called acceleration-roll. The faster the particle moves along the $x_{l}$ axis, the longer the spiral step is.

Like a steel spring that contains an elastic wave, the track in the acceleration-roll in Fig.1(b) can be described by a wave function whose phase changes $2 \pi$ for one spiral step. Apparently, this wave is just the de Broglie's matter wave for electrons, protons or quarks, etc.

Theorem: the acceleration-roll bears matter wave.

$$
\begin{equation*}
\psi=\exp \left(i \frac{\beta}{c^{3}} \int_{0}^{x}\left(u_{1} d x_{1}+u_{2} d x_{2}+u_{3} d x_{3}+u_{4} d x_{4}\right)\right) . \tag{8}
\end{equation*}
$$

Proof: See ref. [28,30].

Depending on the particle under investigation, this wave function may have different explanations. If the $\beta$ is replaced by the Planck constant, the wave function of electrons is given by

$$
\begin{align*}
& \text { assume: } \quad \beta=\frac{m c^{3}}{\hbar}  \tag{9}\\
& \psi=\exp \left(\frac{i}{\hbar} \int_{0}^{x}\left(m u_{1} d x_{1}+m u_{2} d x_{2}+m u_{3} d x_{3}+m u_{4} d x_{4}\right)\right)
\end{align*}
$$

where $m u_{4} d x_{4}=-E d t$, it strongly suggests that the wave function is just the de Broglie's matter wave $[4,5,6]$.

Considering another explanation to $\psi$ for planets in the solar system, no Planck constant can be involved. But, in a many-body system with the total mass $M$, the data analysis [28] tells us that the ultimate acceleration can be rewritten in terms of Planck-constant-like constant $h$ as

$$
\begin{align*}
& \text { assume: } \quad \beta=\frac{c^{3}}{h M}  \tag{10}\\
& \psi=\exp \left(\frac{i}{h M} \int_{0}^{x}\left(u_{1} d x_{1}+u_{2} d x_{2}+u_{3} d x_{3}+u_{4} d x_{4}\right)\right)
\end{align*}
$$

The constant $h$ will be determined by experimental observations. This paper will show that this wave function is applicable to several many-body systems in the solar system, the wave function is called the acceleration-roll wave.

Tip: actually, ones cannot get to see the acceleration-roll of a particle in the relativistic space-time $\left(x_{1}, x_{2}, x_{3}, x_{4}=i c t\right)$; only get to see it in the cylinder coordinate system $\left(x_{1}, x_{4}=i c t, x_{5}=i R\right)$.

## 3. How to determine the ultimate acceleration

In Bohr's orbit model for planets or satellites, as shown in Fig.2, the circular quantization condition is given in terms of relativistic matter wave in gravity by

$$
\left.\begin{array}{c}
\frac{\beta}{c^{3}} \oint_{L} v_{l} d l=2 \pi n  \tag{11}\\
v_{l}=\sqrt{\frac{G M}{r}}
\end{array}\right\} \Rightarrow \sqrt{r}=\frac{c^{3}}{\beta \sqrt{G M}} n ; n=0,1,2, \ldots
$$



Fig. 2 A planet 2D orbit around the sun, an acceleration-roll winding around the planet.
$<$ Clet2020 Script $>/ /[26]$
int i,j,k; double r,rot,x,y,z,D[20],F[20],S[200]; int main() \{SetViewAngle("temp0,theta60,phi-30");
DrawFrame(FRAME_LINE, 1,0xafffaf);r=80;Spiral(); TextHang(r,-r,0,"acceleration-roll");
r=110;TextHang(r,0,0,"x");TextHang(0,r,0,"y");TextHang(0,0,r,"z");\}
Spiral ()$\{\mathrm{r}=80 ; \mathrm{j}=10 ; \mathrm{rot}=\mathrm{j} / \mathrm{r} ; \mathrm{k}=2 * \mathrm{PI} /$ rot +1 ;
for(i=0;i<k;i+=1) $\{\mathrm{D}[0]=x ; \mathrm{D}[1]=\mathrm{y} ; \mathrm{D}[2]=\mathrm{z} ; \mathrm{D}[6]=\mathrm{x} ; \mathrm{D}[7]=\mathrm{y} ; \mathrm{D}[8]=\mathrm{r}$;
$x=r * \cos ($ rot $* i) ; y=r * \sin ($ rot $* i) ; z=0 ; i f(i==0)$ continue;
SetPen(2,0x00);F[0]=D[0];F[1]=D[1];F[2]=x;F[3]=y;Draw("LINE, 0,2,XY,",F);SetPen(1,0xff0000);
$\mathrm{D}[3]=\mathrm{x} ; \mathrm{D}[4]=\mathrm{y} ; \mathrm{D}[5]=\mathrm{z} ; \mathrm{D}[9]=40 ; \mathrm{D}[10]=10 ; \mathrm{D}[11]=8 ; \mathrm{D}[12]=0 ; \mathrm{D}[13]=360$;
Lattice(SPIRAL,D,S);Plot("POLYLINE, 0,40,XYZ",S[9]); \}
\} $\# \mathrm{v} 07=$ ? $>\mathrm{A} \# \mathrm{t}$

The solar system, Jupiter's satellites, Saturn's satellites, Uranus' satellites, and Neptune's satellites as five different many-body systems are investigated with the Bohr's orbit model. After fitting observational data as shown in Fig.3, their ultimate accelerations are obtained in Table 1. The predicted quantization blue-lines in Fig.3(a), Fig.3(b), Fig.3(c), Fig.3(d) and Fig.3(e) agree well with experimental observations for those inner constituent planets or satellites.


Fig. 3 The orbital radii are quantized for inner constituents. (a) the solar system with $h=4.574635 \mathrm{e}-16\left(\mathrm{~m}^{2} \mathrm{~s}^{-1} \mathrm{~kg}^{-1}\right)$. The relative error is less than $3.9 \%$. (b) the Jupiter system with $h=3.531903 \mathrm{e}-16\left(\mathrm{~m}^{2} \mathrm{~s}^{-1} \mathrm{~kg}^{-1}\right)$. Metis and Adrastea are assigned the same quantum number for their almost same radius. The relative error is less than $1.9 \%$. (c) the Saturn system with $h=6.610920 \mathrm{e}-16\left(\mathrm{~m}^{2} \mathrm{~s}^{-1} \mathrm{~kg}^{-1}\right)$. The relative error is less than $1.1 \%$. (d) the Uranus system with $h=1.567124 \mathrm{e}-16\left(\mathrm{~m}^{2} \mathrm{~s}^{-1} \mathrm{~kg}^{-1}\right) . n=0$ is assigned to Uranus. The relative error is less than $2.5 \%$. (e) the Neptune system with $h=1.277170 \mathrm{e}-16\left(\mathrm{~m}^{2} \mathrm{~s}^{-1} \mathrm{~kg}^{-1}\right) . n=0$ is assigned to Neptune. The relative error is less than $0.17 \%$.

Table 1 Planck-constant-like constant $h, \mathrm{~N}$ is constituent particle number with smaller orbital inclination.

| system | N | $M / M_{\text {earth }}$ | $\beta(\mathrm{m} / \mathrm{s} 2)$ | $h(\mathrm{~m} 2 \mathrm{~s}-1 \mathrm{~kg}-1)$ | Prediction |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Solar planets | 9 | 333000 | $2.961520 \mathrm{e}+10$ | $4.574635 \mathrm{e}-16$ | Fig.3(a) |
| Jupiter' satellites | 7 | 318 | $4.016793 \mathrm{e}+13$ | $3.531903 \mathrm{e}-16$ | Fig.3(b) |
| Saturn's satellites | 7 | 95 | $7.183397 \mathrm{e}+13$ | $6.610920 \mathrm{e}-16$ | Fig.3(c) |
| Uranus' satellites | 18 | 14.5 | $1.985382 \mathrm{e}+15$ | $1.567124 \mathrm{e}-16$ | Fig.3(d) |
| Neptune 's satellites | 7 | 17 | $2.077868 \mathrm{e}+15$ | $1.277170 \mathrm{e}-16$ | Fig.3(e) |

Besides every $\beta$, our interest shifts to the constant $h$ in Table 1 , which is defined as

$$
\begin{equation*}
h=\frac{c^{3}}{M \beta} \Rightarrow \sqrt{r}=h \sqrt{\frac{M}{G}} n . \tag{12}
\end{equation*}
$$

In a many-body system with a total mass of $M$, a constituent particle has the mass of $m$ and moves at the speed of $v$, it is easy to find that the wavelength of de Broglie's matter wave should be modified for planets and satellites as

$$
\begin{equation*}
\lambda_{d_{-}-B r o g l i e}=\frac{2 \pi \hbar}{m v} \Rightarrow \text { modify } \Rightarrow \lambda=\frac{2 \pi h M}{v} \tag{13}
\end{equation*}
$$

where $h$ is a Planck-constant-like constant. Usually, the total mass $M$ is approximately equal to the central-star's mass. It is found that this modified matter wave works for quantizing orbits correctly in Fig. 3 [28,29]. The key point is that the various systems have almost the same Planck-constant-like constant $h$ in Table 1 with a mean value of $3.51 \mathrm{e}^{-16} \mathrm{~m}^{2} \mathrm{~s}^{-1} \mathrm{~kg}^{-1}$, at least having the same magnitude! The acceleration-roll wave is a generalized matter wave on a planetary scale.

In Fig.3(a), the blue straight line expresses the linear regression relation among the Sun, Mercury, Venus, Earth and Mars, their quantization parameters are $h M=9.098031 \mathrm{e}+14\left(\mathrm{~m}^{2} / \mathrm{s}\right)$. The ultimate acceleration is fitted out to be $\beta=2.961520 \mathrm{e}+10\left(\mathrm{~m} / \mathrm{s}^{2}\right)$. Where, $n=3,4,5, .$. were assigned to solar planets, the sun was assigned a quantum number $n=0$ because the sun is in the central state.

## 4. Optical model of the central state

The acceleration-roll wave as the relativistic matter wave generalized in gravity is given by

$$
\begin{equation*}
\psi=\exp \left(\frac{i}{h M} \int_{0}^{x} v_{l} d l\right) ; \quad \lambda=\frac{2 \pi h M}{v_{l}}=\frac{2 \pi c^{3}}{v_{l} \beta} . \tag{14}
\end{equation*}
$$

In a central state $n=0$, if the coherent length of the acceleration-roll wave is long enough, its head may overlap with its tail when the particle moves in a closed orbit in space-time, as shown in Fig.4, the interference of the acceleration-roll wave between its head and tail will occur in the overlapping zone. The overlapped wave is given by

$$
\begin{align*}
& \psi(r)=1+e^{i \delta}+e^{i 2 \delta}+\ldots+e^{i(N-1) \delta}=\frac{1-\exp (i N \delta)}{1-\exp (i \delta)} \\
& \delta(r)=\frac{1}{h M} \oint_{L}\left(v_{l}\right) d l=\frac{2 \pi \omega r^{2}}{h M}=\frac{2 \pi \beta \omega r^{2}}{c^{3}} \tag{15}
\end{align*}
$$

where $N$ is the overlapping number which is determined by the coherent length of the acceleration-roll wave, $\delta$ is the phase difference after one orbital motion, $\omega$ is the angular speed of the solar rotation. The above equation is a multi-slit interference formula in optics, for a larger $N$ it is called the Fabry-Perot interference formula.


Fig. 4 The head of the acceleration-roll wave may overlap with its tail.

The acceleration-roll wave function $\psi$ needs a further explanation. In quantum mechanics, $|\psi|^{2}$ equals to the probability of finding an electron due to Max Burn's explanation; in astrophysics, $|\psi|^{2}$ equals to the probability of finding a nucleon (proton or neutron) averagely on an astronomic scale, because all mass is mainly made of nucleons, we have

$$
\begin{equation*}
|\psi|^{2} \propto \text { nucleon_density } \tag{16}
\end{equation*}
$$

It follows from the multi-slit interference formula that the interference intensity at maxima is proportional to $N^{2}$, that is

$$
\begin{equation*}
N^{2}=\frac{\left|\psi(0)_{\text {multi-wavelet }}\right|^{2}}{\left|\psi(0)_{\text {one-wavelet }}\right|^{2}} . \tag{17}
\end{equation*}
$$

What matter plays the role of "one-wavelet" in the solar core or earth core? We choose airvapor at the sea level on the earth's surface as the "reference matter: one-wavelet". Thus, the overlapping number $N$ is estimated by

$$
\begin{equation*}
N^{2}=\frac{\left|\psi(0)_{\text {multi-wavelet }}\right|^{2}}{\left|\psi(0)_{\text {one-wavelet }}\right|^{2}} \approx \frac{\text { core_nucleon_density }_{r=0}}{\text { air_}_{\text {ond }} \text { vapor_density }} r_{r=\text { sea }} \quad . \tag{18}
\end{equation*}
$$

Although today there is no air-vapor on the solar surface, does not hinder to select it as the reference matter. The solar core has a maximum density of $1.5 \mathrm{e}+5 \mathrm{~kg} / \mathrm{m}^{3}[31]$, comparing to the air-vapor density of $1.29 \mathrm{~kg} / \mathrm{m}^{3}$ at the sea level on the earth, the solar overlapping number $N$ is estimated as $N=341$. The earth's core density is $5.53 \mathrm{e}+3 \mathrm{~kg} / \mathrm{m}^{3}$, the earth's overlapping number $N$ is estimated as $N=65$.

For the Sun, Earth and Mars, their central densities and their reference matter density are
given in Table 2. Thus, their overlapping numbers are estimated also in this table.
Sun's angular speed at the equator is known as $\omega=2 \pi /(25.05 * 24 * 3600)$, unit: $\mathrm{s}^{-1}$. Its mass $1.9891 \mathrm{e}+30(\mathrm{~kg})$, radius $6.95 \mathrm{e}+8(\mathrm{~m})$, mean density $1408\left(\mathrm{~kg} / \mathrm{m}^{3}\right)$, the solar core has a maximum density of $1.5 \mathrm{e}+5 \mathrm{~kg} / \mathrm{m}^{3}$ [31], the ultimate acceleration $\beta=2.961520 \mathrm{e}+10\left(\mathrm{~m} / \mathrm{s}^{2}\right)$, the constant $h M=9.100745 \mathrm{e}+14\left(\mathrm{~m}^{2} / \mathrm{s}\right)$. According to the $N=341$, the matter distribution of the $|\psi|^{2}$ is calculated in Fig.5, it agrees well with the general description of the sun's interior. The radius of the sun is calculated to be $r=7 \mathrm{e}+8(\mathrm{~m})$ with a relative error of $0.72 \%$ in Fig.5, which indicates that the sun radius strongly depends on the sun's self-rotation.

Table 2 Estimating the overlapping number $N$ by comparing solid core to reference matter, regarding protons and neutrons as basis particles.

| object | Solid core, <br> density $\left(\mathrm{kg} / \mathrm{m}^{3}\right)$ | Reference matter, <br> density $\left(\mathrm{kg} / \mathrm{m}^{3}\right)$ | Overlapping <br> number $N$ | $\beta$ <br> $\left(\mathrm{m} / \mathrm{s}^{2}\right)$ |
| :---: | :---: | :---: | :---: | :---: |
| Sun | $1.5 \mathrm{e}+5(\mathrm{max})$. | 1.29 (vapor above the sea) | 341 | $2.961520 \mathrm{e}+10$ |
| Earth | 5530 | 1.29 (vapor above the sea) | 65 | $1.377075 \mathrm{e}+14$ |
| Mars | 3933.5 | 1.29 (vapor above the sea) | 55 | $2.581555 \mathrm{e}+15$ |
| Jupiter | 1326 |  |  | $4.016793 \mathrm{e}+13$ |
| Saturn | 687 |  |  | $7.183397 \mathrm{e}+13$ |
| Uranus | 1270 |  |  | $1.985382 \mathrm{e}+15$ |
| Neptune | 1638 |  |  | $2.077868 \mathrm{e}+15$ |
| Alien-planet | 5500 | 1.29 (has water on the surface) | 65 |  |



Fig. 5 The matter distribution $|\psi| 2$ around the Sun has been calculated in the radius direction.

[^0]
## 5. Earth's central state and space debris distribution

Appling the acceleration-roll wave to the Moon, as illustrated in Fig.6(a), the Moon has been assigned a quantum number of $n=2$ in the author's early study [28]. According to the quantum condition, the ultimate acceleration is fitted out to be $\beta=1.377075 \mathrm{e}+14\left(\mathrm{~m} / \mathrm{s}^{2}\right)$ in the earth system. Another consideration is to take the quasi-satellite's perigee into account, for the moon and 2004_GU9 etc., as shown in Fig.6(b), but this consideration requires further understanding of its five quasi-satellites [28].


Fig. 6 Orbital quantization for moon and quasi-satellites to the Earth, $H=h M$.

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char $\operatorname{str}[200]$;int $\mathrm{i}, \mathrm{j}, \mathrm{k}, \mathrm{N}, \mathrm{nP}[10]$; double $\mathrm{x}, \mathrm{y}, \mathrm{z}, \mathrm{M}, \mathrm{r}_{\text {_unit }}, \mathrm{a}, \mathrm{b}, \mathrm{B}, \mathrm{H}, \mathrm{r}_{2}$ ave[20], $\mathrm{dP}[10], \mathrm{D}[1000$ ];
double orbit $[10]=\{0,2.57,0$,$\} ; double \mathrm{e}[10]=\{0,0.0549,0,0,0,0, \overline{0}, 0,0,0$,$\} ;$
int qn $[10]=\{0,2,3,4,5,6,7,8,9 ., 10$,$\} ; char Stars [100]=\{$ "Earth;Moon;" $\}$;
int main() $\{\mathrm{N}=2 ; \mathrm{M}=5.97237 \mathrm{E} 24 ; \mathrm{r}$ unit=1.495978707e8
for $(\mathrm{i}=0 ; \mathrm{i}<\mathrm{N} ; \mathrm{i}+=1)\{\mathrm{x}=\operatorname{orbit}[\mathrm{i}] ; \mathrm{y}=\mathrm{e}[\mathrm{i}] ; \mathrm{z}=\mathrm{x} *(1+\operatorname{sqrt}(1-\mathrm{y} * \mathrm{y})) / 2 ; \mathrm{r}$ ave[i]= $\mathrm{z} ; / /$ average_radius
$\mathrm{D}[\mathrm{i}+\mathrm{i}]=\mathrm{qn}[\mathrm{i}] ; \mathrm{D}[\mathrm{i}+\mathrm{i}+1]=\operatorname{sqrt}(\mathrm{z}) ;\}$
DataJob("REGRESSION,2",D, dP$) ; \mathrm{b}=\mathrm{dP}[0] ; \mathrm{a}=\mathrm{dP}[1]$;
SetAxis(X_AXIS, $0,0,3, " n ; 0 ; 1 ; 2 ; 3 ; ") ;$
SetAxis(Y AXIS, $0,0,3, " \# i f \# r s r \# t($ average radius unit:0.001 AU); $0 ; 1 ; 2 ; 3 ; "$ )
DrawFramé(0x0166,1,0xafffaf); Polyline(N,D);
SetPen(2,0xff0000); Plot("OVALFILL,0,2,XY,3,3,",D);
for $(\mathrm{i}=0 ; \mathrm{i}<\mathrm{N} ; \mathrm{i}+=1)\{\mathrm{nP}[0]=\mathrm{TAKE} ; \mathrm{nP}[1]=\mathrm{i} ; \operatorname{TextJob}(\mathrm{nP}, \operatorname{Stars}, \operatorname{str}) ; \mathrm{x}=\mathrm{qn}[\mathrm{i}]+0.2 ; \mathrm{y}=\operatorname{sqrt}(\operatorname{orbit}[\mathrm{i}])-0.05 ; \operatorname{TextHang}(\mathrm{x}, \mathrm{y}, 0, \mathrm{str}) ;\}$
$\mathrm{x}=\mathrm{GRAVITYC} \mathrm{M}^{*} \mathrm{r}$ unit; $\mathrm{z}=\operatorname{sqrt}(\mathrm{x}) ; \mathrm{H}=\mathrm{z}^{*} \mathrm{a} ; \mathrm{B}=-\mathrm{z} * \mathrm{~b}$;
TextAt(100,450,"\#if $\bar{H} \# t=\%$ e $\left.\# i f B \# t=\% e^{\prime}, H, B\right) ;$
for $(\mathrm{i}=0 ; \mathrm{i}<\mathrm{N} ; \mathrm{i}+=1)\{\mathrm{y}=\mathrm{b}+\mathrm{a} * \mathrm{qn}[\mathrm{i}] ; \mathrm{D}[\mathrm{i}+\mathrm{i}]=\mathrm{qn}[\mathrm{i}] ; \mathrm{D}[\mathrm{i}+\mathrm{i}+1]=\mathrm{y} ;\}$
SetPen(1,0x0000ff);Polyline(N,D,0.5,2.2,"quantization");//check
\}\} $\# \mathrm{v} 07=$ ? $>\mathrm{A} \# \mathrm{t}$

The earth's angular speed is known as $\omega=2 \pi /(24 * 3600)$, unit s ${ }^{-1}$. Its mass $5.97237 \mathrm{e}+24(\mathrm{~kg})$, radius $6.371 \mathrm{e}+6(\mathrm{~m})$, core density $5530\left(\mathrm{~kg} / \mathrm{m}^{3}\right)$, the ultimate acceleration $\beta=1.377075 \mathrm{e}+14\left(\mathrm{~m} / \mathrm{s}^{2}\right)$, the constant $h M=1.956611 \mathrm{e}+11\left(\mathrm{~m}^{2} / \mathrm{s}\right)$.

We have estimated that the wave overlapping number in the central state of the earth is $N=65$, the matter distribution $|\psi|^{2}$ in radius direction is calculated as shown in Fig.7(a), where the self-rotation near its equator has the period of 24 hours:

$$
\begin{equation*}
\delta(r)=\frac{1}{h M} \oint_{L}\left(v_{l}\right) d l=\frac{2 \pi r}{h M} \omega r=\frac{2 \pi \beta \omega r^{2}}{c^{3}} . \tag{19}
\end{equation*}
$$

The matter distribution has a central maximum at the earth's heart, which gradually decreases to zero near the earth surface, then rises the secondary peaks and attenuates down off. The radius of the earth is calculated to be $r=6.4328 \mathrm{e}+6(\mathrm{~m})$ with a relative error of $0.86 \%$ using the interference of its acceleration-roll wave. Space debris over the atmosphere has a complicated evolution $[7,8]$, and has itself speed

$$
\begin{equation*}
v_{l}=\sqrt{\frac{G M}{r}} ; \quad \delta(r)=\frac{1}{h M} \oint_{L}\left(v_{l}\right) d l=\frac{\beta}{c^{3}} \oint_{L}\left(v_{l}\right) d l=\frac{2 \pi \beta}{c^{3}} \sqrt{G M r} . \tag{20}
\end{equation*}
$$

The secondary peaks over the atmosphere up to 2000 km altitude are calculated in Fig.7(b) which agree well with the space debris observations [16]; the peak near 890 km altitude is due principally to the January 2007 intentional destruction of the Feng Yun-1C weather spacecraft, while the peak centered at approximately 770 km altitude was created by the February 2009 accidental collision of Iridium 33 (active) and Cosmos 2251 (derelict) communication spacecraft $[16,18]$. The observations based on the incoherent scattering radar EISCAT ESR located at $78^{\circ} \mathrm{N}$ in Jul. 2006 and in Oct. 2015 [21,22,23] are respectively shown in Fig.7(c) and (d). This prediction of secondary peaks also agrees well with other space debris observations [24,25].
 interference of its acceleration-roll wave; (b) The prediction of the space debris distribution up to 2000 km altitude; (c) The pace debris distribution in Jul. 2006, Joint observation based on the incoherent scattering radar EISCAT ESR located at $78^{\circ} \mathrm{N}$ [21]; (d) The space debris distribution in Oct. 2015, Joint observation based on the incoherent scattering radar EISCAT ESR located at $78^{\circ} \mathrm{N}$ [21].

[^1]$<$ Clet2020 Script>//[26]
int i,j, k,m,n,N,nP[10]; double H,B,M,v r,r,AU,r unit, X, y, Z,delta,D[10], S[10000];
double rs,rc,rot,a,b,atm height,p,T,R1, $\overline{\mathrm{R}} 2, \mathrm{R} 3$; char str[100]; int
Debris[96] $=\{110,0,237,0,287,0,317,2,320,1,357,5,380,1,387,4,420,2,440,3,454,14,474,9,497,45,507,26,527,19,557,17,597,34,63$
4,37,664,37,697,51,727,55,781,98,808,67,851,94,871,71,901,50,938,44,958,44,991,37,1028,21,1078,17,1148,10,1202,9,1225,6,
$1268,12,1302,9,1325,5,1395,7,1395,18,1415,36,1429,12,1469,22,1499,19,1529,9,1559,5,1656,4,1779,1,1976,1$,
$\operatorname{main}()\{\mathrm{k}=80 ; \mathrm{rs}=6.378 \mathrm{e} 6 ; \mathrm{rc}=0 ; \mathrm{atm}$ height $=1.5 \mathrm{e} 5 ; \mathrm{n}=0 ; \mathrm{N}=65$;
$\mathrm{H}=1.956611 \mathrm{e} 11 ; \mathrm{M}=5.97237 \mathrm{e} 24 ; \mathrm{A} \overline{\mathrm{U}}=1.496 \mathrm{E} 11 ;$ r_unit $=1 \mathrm{e} 4$;
rot $=2 * \mathrm{PI} /(24 * 60 * 60) ; / /$ angular speed of the Earth
$\mathrm{b}=\mathrm{PI} /\left(2 * \mathrm{PI}^{*} \mathrm{rot} * \mathrm{rs} * \mathrm{rs} / \mathrm{H}\right) ; \mathrm{R} 1=\mathrm{rs} / \mathrm{r}$ _unit;R2=(rs+atm_height)/r unit;R3=(rs+2e6)/r_unit;
for( $\mathrm{i}=\mathrm{R} 2 ; \mathrm{i}<\mathrm{R} 3 ; \mathrm{i}+=1)\left\{\mathrm{r}=\mathrm{abs}(\mathrm{i}) * \mathrm{r}\right.$ - unnit; delta=2*PI*sqrt(GRAV$\left.I T Y C * \mathrm{M}^{*} \mathrm{r}\right) / \mathrm{H}$;
y=SumJob("SLIT_ADD,@N,@-̄elta",D); y=1e3*y/(N*N);// visualization scale:1000
if $(\mathrm{y}>1) \mathrm{y}=1 ; \mathrm{S}[\mathrm{n}]=\mathrm{i} ; \mathrm{S}[\mathrm{n}+1]=\mathrm{y} ; \mathrm{n}+=2 ;\}$
SetAxis(X_AXIS,R1,R1,R3,"altitude; r\#sds\#t;500;1000;1500;2000km ;");
SetAxis(Y_AXIS, $0,0,1$,"\#if| $\mid \#$ su2\#t; $0 ; ~ ; 1 \mathrm{e}-3 ; ") ;$ DrawFrame(FRAME SCALE, 1,0xafffaf); $\mathrm{x}=\mathrm{R} 1+(\mathrm{R} 3-\mathrm{R} 1) / 5$;
SetPen(1,0xff0000);Polyline(n/2,S,x,0.8,"\#if $\Psi \Psi \mid \#$ su2\#t (density, prediction)");
for $(\mathrm{i}=0 ; \mathrm{i}<48 ; \mathrm{i}+=1)\{\mathrm{S}[\mathrm{i}+\mathrm{i}]=\mathrm{R} 1+(\mathrm{R} 3-\mathrm{R} 1) * \operatorname{Debris}[\mathrm{i}+\mathrm{i}] / 2000 ; \mathrm{S}[\mathrm{i}+\mathrm{i}+1]=\operatorname{Debris}[\mathrm{i}+\mathrm{i}+1] / 300 ;\}$
SetPen(1,0x0000ff);Polyline(48,S,x,0.7,"Space debris (2018, observation) "); \}\#v07=?>A\#t

## 6. Sunspot cycle

The coherence length of waves is usually mentioned but the coherence width of waves is rarely discussed in quantum mechanics, simply because the latter is not a matter for electrons, nucleon, or photons, but it is a matter in astrophysics. The analysis of observation data tells us that on the planetary scale, the coherence width of acceleration roll waves can be extended to 1000 kilometers or more, as illustrated in Fig.8(a), the overlap may even occur in the width direction, thereby bringing new aspects to wave interference.

In the solar convective zone, adjacent convective arrays form a top-layer flow, a middlelayer gas, and a ground-layer flow, similar to the concept of molecular current in electromagnetism. Considering one convective ring at the equator as shown in Fig.8(b), there is an apparent velocity difference between the top-layer flow and the middle-layer gas, where their acceleration-roll waves are denoted respectively by


Fig. 8 (a) Illustration of overlapping in the coherent width direction. (b) In convective rings at the equator, the speed difference causes a beat frequency.

[^2]$<$ Clet2020 Script>//Clet is a C compiler[26]
double beta,H,M,N,dP[20],D[2000],r,rs,rot, x,y,v1,v2,K1,K2,T1,T2,T,Lamda,L; int i,j,k;
int main() $\{$ beta $=2.961520 \mathrm{e} 10 ; \mathrm{H}=$ SPEEDC*SPEEDC*SPEEDC/beta;
$\mathrm{M}=1.9891 \mathrm{E} 30 ; \mathrm{rs}=6.95 \mathrm{e} 8 ; \mathrm{rot}=2 * \mathrm{PI} /(25.05 * 24 * 3600) ; \mathrm{vl}=\mathrm{rot} * \mathrm{rs} ; \mathrm{K} 1=\mathrm{v} 1 * \mathrm{v} 1 / 2 ; / / \mathrm{T} 1=2 * \mathrm{PI} * \mathrm{H} / \mathrm{K} 1$;
$\mathrm{v} 2=6100 ; / / 0.7346 *$ sqrt(BOLTZMANN*5700/MP) $+0.2485 *$ sqrt(BOLTZMANN*5700/(MP+MP));
$\mathrm{K} 2=\mathrm{v} 2 * \mathrm{v} 2 / 2 ; \mathrm{T} 2=2 * \mathrm{PI} * \mathrm{H} /(\mathrm{K} 2-\mathrm{K} 1) ; \mathrm{T}=\mathrm{T} 2 / 24 * 3600 * 365.2422$;
Lamda $=2 * \mathrm{PI} * \mathrm{H} /(\mathrm{v} 2-\mathrm{v} 1) ; \mathrm{L}=2 * \mathrm{PI} * \mathrm{rs} / \mathrm{Lamda}$;
SetViewAngle("temp0,theta60,phi-60");
DrawFrame(FRAME_LINE,1,0xafffaf);Overlook("2,1,60", D);
TextAt(10,10,"v1=\% $\%$, v2 $=\% \mathrm{~d}, \mathrm{~T}=\% .2 \mathrm{f} y$, $\# \mathrm{n} \lambda=\% \mathrm{e}, \mathrm{L}=\% \mathrm{f}^{\prime}, \mathrm{v} 1, \mathrm{v} 2, \mathrm{~T}$, Lamda,L);
SetPen(1,0x4f4fff); for (i=0;i<18;i+=1) \{v1=i*2*PI/18; x=70* $\cos (v 1) ; y=70 * \sin (v 1) ; \operatorname{Ring}() ;\}$
SetPen(2,0xff0000);Draw("ARROW,0,2,XYZ,15","80,0,0,80,60,0");
TextHang(100,20,0,"top-layer $\omega \#$ sdl\#t"); SetPen(2,0x0000ff);
Draw("ARROW,0,2,XYZ,15","70,0,0,70,60,0");
TextHang(50,60,0,"center $\omega \#$ sd2\#t");TextHang(140,-30,0," $\omega \#$ sdbeat\#t= $\omega \#$ sd1\#t- $\omega \# s d 2 \# t ") ;$
Ring() $\{\mathrm{k}=0 ; \mathrm{N}=20 ; \mathrm{r}=10$;
for $(\mathrm{j}=0 ; \mathrm{j}<\mathrm{N}+2 ; \mathrm{j}+=1)\{\mathrm{k}=\mathrm{j}+\mathrm{j}+\mathrm{j}$; v2=j*2*PI/N;
$\mathrm{D}[\mathrm{k}]=\mathrm{x}+\mathrm{r} * \cos (\mathrm{v} 2) ; \mathrm{D}[\mathrm{k}+1]=\mathrm{y}+\mathrm{r} * \sin (\mathrm{v} 2) ; \mathrm{D}[\mathrm{k}+2]=0 ;\}$
Plot("POLYLINE,4,22,XYZ, 8",D);\}
\#v07=?>A
\[

$$
\begin{align*}
& \psi=\psi_{\text {top }}+C \psi_{\text {middle }} \\
& \psi_{\text {top }}=\exp \left[\frac{i \beta}{c^{3}} \int_{L}\left(v_{1} d l+\frac{-c^{2}}{\sqrt{1-v_{1}^{2} / c^{2}}} d t\right)\right]  \tag{21}\\
& \psi_{\text {middle }}=\exp \left[\frac{i \beta}{c^{3}} \int_{L}\left(v_{2} d l+\frac{-c^{2}}{\sqrt{1-v_{2}^{2} / c^{2}}} d t\right)\right]
\end{align*}
$$
\]

Their interference in the coherent width direction leads to a beat phenomenon

$$
\begin{align*}
& |\psi|^{2}=\left|\psi_{\text {top }}+C \psi_{\text {middle }}\right|^{2}=1+C^{2}+2 C \cos \left[\frac{2 \pi}{\lambda_{\text {beat }}} \int_{L} d l-\frac{2 \pi}{T_{\text {beat }}} t\right] \\
& \frac{2 \pi}{T_{\text {beat }}}=\frac{\beta}{c^{3}}\left(\frac{c^{2}}{\sqrt{1-v_{1}^{2} / c^{2}}}-\frac{c^{2}}{\sqrt{1-v_{2}^{2} / c^{2}}}\right) \simeq \frac{\beta}{c^{3}}\left(\frac{v_{1}^{2}}{2}-\frac{v_{2}^{2}}{2}\right)  \tag{22}\\
& \frac{2 \pi}{\lambda_{\text {beat }}}=\frac{\beta}{c^{3}}\left(v_{1}-v_{2}\right) ; \quad V=\frac{\lambda_{\text {beat }}}{T_{\text {beat }}}=\frac{1}{2}\left(v_{1}+v_{2}\right)
\end{align*}
$$

Their speeds are calculated as

$$
\begin{align*}
& v_{1} \approx 6200(\mathrm{~m} / \mathrm{s}) \quad(\approx \text { observed in Evershed flow })  \tag{23}\\
& v_{2}=\omega r_{\text {middle }}=2017(\mathrm{~m} / \mathrm{s}) \quad(\text { solar rotation })
\end{align*}
$$

Where, regarding Evershed flow as the eruption of the top-layer flow, about $6 \mathrm{~km} / \mathrm{s}$ speed was reported [31]. Alternatively, the top-layer speed $v_{1}$ also can be calculated in terms of thermodynamics, to be $v_{1}=6244(\mathrm{~m} / \mathrm{s})$ [28]. Here using $v_{1}=6100(\mathrm{~m} / \mathrm{s})$, their beat period $T_{\text {beat }}$ is calculated to be a very remarkable value of 10.93 (years), in agreement with the sunspot cycle value (say, mean 11 years).

$$
\begin{equation*}
T_{b e a t} \simeq \frac{4 \pi c^{3}}{\beta\left(v_{1}^{2}-v_{2}^{2}\right)}=10.93(\text { years }) \tag{24}
\end{equation*}
$$

The relative error to the mean 11 years is $0.6 \%$ for the beat period calculation using the acceleration-roll waves. This beat phenomenon turns out to be a nucleon density oscillation that undergoes to drive the sunspot cycle evolution. The beat wavelength $\lambda_{\text {beat }}$ is too long to observe, only the beat period is easy to be observed. As shown in Fig.9, on the solar surface, the equatorial circumference $2 \pi r$ only occupies a little part of the beat wavelength, what we see is
the expansion and contraction of the nucleon density.

$$
\begin{equation*}
\frac{2 \pi r}{\lambda_{\text {beat }}}=0.0031 . \tag{25}
\end{equation*}
$$

This nucleon density oscillation is understood as a new type of nuclear reaction on an astronomic scale.


Fig. 9 The equatorial circumference $2 \pi r$ only occupies a little part of the beat wavelength, what we see is the expansion and contraction of the nucleon density.

In the above calculation, although this seems to be a rough model, there is an obvious correlation between solar radius, solar rotation, solar density, ultimate acceleration, and Planck-constant-like constant $h$.

## 7. Atmospheric circulation

Consider an acceleration-roll wave $\psi_{A}$ in the earth shell at the latitude angle $A$, it will interfere with its neighbor waves within its coherent width. Because the earth shell mainly consists of dense matter, their mutual cascade-interference will cause the acceleration roll wave to have the same phase at the same longitude, so that the acceleration roll wave $\psi_{A}$ should equal to the $\psi_{\text {equator }}$ at the same longitude, as shown in Fig.10(a). This conclusion is supported by the spherical symmetry of the Earth's density distribution, that is.


Fig. 10 (a) Mutual cascade-interference will lead to the acceleration-roll wave having the same phase. (b) Jupiter
<Clet2020 Script>// [26]
double dP[20],D[2000], r, v1,v2,K1,K2; int i,j,k,N;
int main() \{SetViewAngle("temp0, theta60,phi-60");
DrawFrame(FRAME LINE,1,0xafffaf);Overlook("2,1,60", D);
for $(\mathrm{i}=0 ; \mathrm{i}<180 ; \mathrm{i}+=15)\{\mathrm{k}=0 ; \mathrm{K} 1=0 ; \mathrm{K} 2=\mathrm{i} ; \operatorname{Grid}() ;\}$
for $(\mathrm{i}=0 ; \mathrm{i}<180 ; \mathrm{i}+=15)\{\mathrm{k}=1 ; \mathrm{K} 1=\mathrm{i} ; \mathrm{K} 2=0 ; \operatorname{Grid}() ;\}$
SetPen(3,0xff0000); k=1; K1=60; K2=0;Grid();K1=90; K2=0; Grid();
//SetPen(3,0x4f4fff); k=1; K1=55; K2=0;Grid();K1=65; K2=0; Grid();
TextHang(40,40,50,"\#ify\#sdA\#t"); TextHang(50,40,0,"\#ify\#sdequator\#t");
\}
Grid() $\{\mathrm{N}=50 ; \mathrm{K} 1 *=\mathrm{PI} / 180 ; \mathrm{K} 2 *=\mathrm{PI} / 180 ; \quad \mathrm{r}=60$;
if(k==0) $\{\mathrm{v} 1=2 * \mathrm{PI} / \mathrm{N} ; \mathrm{v} 2=0 ;\}$ else $\{\mathrm{v} 1=0 ; \mathrm{v} 2=2 * \mathrm{PI} / \mathrm{N} ;\}$
for $(\mathrm{j}=0 ; \mathrm{j}<=\mathrm{N} ; \mathrm{j}+=1)\{\mathrm{k}=\mathrm{j}+\mathrm{j}+\mathrm{j}$;
$\mathrm{D}[\mathrm{k}]=\mathrm{r} * \sin (\mathrm{~K} 1) * \cos (\mathrm{~K} 2) ; \mathrm{D}[\mathrm{k}+1]=\mathrm{r} * \sin (\mathrm{~K} 1) * \sin (\mathrm{~K} 2) ; \mathrm{D}[\mathrm{k}+2]=\mathrm{r} * \cos (\mathrm{~K} 1)$;
$\mathrm{K} 1+=\mathrm{v} 1 ; \mathrm{K} 2+=\mathrm{v} 2 ;\}$ Plot("POLYGON,0,50, XYZ,10",D);
\}\#v07=? $>\mathrm{A} \# \mathrm{t}$

On the contrary, in the thin atmosphere, the cascade-interference within the coherence width can be ignored, so the wind and clouds are widely distributed in the sky on a large scale.

Through the coherent width concept, considering the interference between the air $\psi_{A}$ at the latitude angle $A$ and the shell $\psi_{\text {shell }}$ at the same latitude, they are

$$
\begin{align*}
& \psi(r, A)=\psi_{\text {air }}(r, A)+C \psi_{\text {shell }}(r, A)=\psi_{\text {air }}(r, A)+C \psi_{\text {shell_equator }}(r) \\
& T_{\text {beat }} \simeq \frac{4 \pi c^{3}}{\beta\left(v_{\text {shell_equator }}{ }^{2}-v_{\text {air }}{ }^{2}\right)}  \tag{27}\\
& v_{\text {shell_equator }}=\omega r \\
& v_{\text {air }}=\omega r \cos (A)+v_{\text {wind }}+v_{\text {sun_effect }}
\end{align*}
$$

where $C$ represents the coupling constant which relates to their distance and mass fractions, their interference leads to a beat phenomenon. The positive wind denotes the direction from west to east. The air is subjected to the solar radiation which enforces the beat oscillation to run at the period $T_{\text {beat }}=1$ (year) at the latitude angles $A=-23.5^{\circ} \mathrm{N} \sim 23.5^{\circ} \mathrm{N}$ due to the tilt of the earth axis concerning the earth orbital plane. We have known that there is a beat $T_{\text {beat }}=1$ year in the first constructive interference ridge with zero wind at the latitude $A_{1}$, using the above neat period formula we obtain the convictive effect:

$$
\begin{equation*}
v_{\text {sun_effect }}=369.788-\omega r \cos \left(A_{1}\right) ; \quad(\text { units }: m / s) \tag{28}
\end{equation*}
$$

It is not easy to maintain the constructive interference condition for these waves. When the first ridge is at latitude $A_{1}=12^{\circ} \mathrm{N}$, the wind required for maintaining the beat $T_{\text {beat }}=1$ (year) near the latitude $A=12^{\circ} \mathrm{N}$ is calculated as shown in Fig.11(a) (blue line), this wind will be destroyed by destructive interference at higher latitudes. But, the wind will arise up again at the next locations where the waves satisfy the constructive interference condition: at $A=50^{\circ} \mathrm{N}$ location where beat $T_{\text {beal }}=0.5$ (years), and at $A=70^{\circ} \mathrm{N}$ location where beat $T_{\text {beat }}=0.37$ (years) which is the shortest period that the earth can get within the arctic regions. The maximal wind appears most probably at the midst of the first two ridges, about $48 \mathrm{~m} / \mathrm{s}$. Linking all characteristic points in Fig. 11(a) we obtain the predicted wind-curve over the northern hemisphere; this prediction agrees well with the experimental observations at an altitude of $10 \mathrm{~km}(200 \mathrm{hPa}$ ), as shown in Fig. 12.

(a) Calculation of west winds in the northern hemisphere. (b) The atmospheric circulation in the northern hemisphere.

```
<Clet2020 Script>// [26]
double beta,H,M,r,rc, rs, rot,v1,v2, Year,T,Lamda,V,a,b,w,Fmax,N[500],S[500],F[100]; int i, j, k, t, m, n, s, f,Type,x;
int main() {beta=1.377075e+14; H=SPEEDC*SPEEDC*SPEEDC/beta;
M=5.97237e24; rs=6.371e6; rot=2*PI/(24*3600); Year=24*3600*365.2422
Type=1; x=10; if(Type>1) x=-30;
if(Type==1) SetAxis(X_AXIS,0,0,90,"Latitude#n(}\mp@subsup{}{}{\circ}\textrm{N});0;30;60;90;")
else SetAxis(X AXIS,-90,-90,90,"Latitude#n('}\mp@subsup{}{}{\circ}\textrm{N});=90;-60;-30;0;30;60;90;")
SetAxis(Y AXIS,-100,-100,100,"West wind (m/s);-100;-80;-60;-40;-20;0;20;40;60;80;100;");
DrawFramè(0x016a,Type,0xafffaf);//Polyline(2,"-90,0,90,0");
Check(15,k); if(k>24) k=24; if(k<0) k=0;
T=Year/2; Wind(); f=0; Findf(); t=N[m+m]; T=Year; Wind(); f=0; Findf();
SetPen(2,0xff); Polyline(n,N,x,70,"Wind for T#sdbeat#t=1 year"); if(Type>1) Polyline(s,S);
F[0]=N[0];F[1]=N[1]; F[2]=N[m+m]; F[3]=N[m+m+1]; t=(t+F[2])/2;//midst of two ridges
t=t-F[2]+m; Fmax =N[tt+t+1];//TextAt(100,20,"t=%d, Fmax=%f ",t,Fmax);
f=Fmax; Findf(); F[4]=N[m+m];F[5]=N[m+m+1];
T=Year/2; Wind(); f=-Fmax/2; Findf(); t=m;f=Fmax/2; Findf();
SetPen(2,0x80ff00); Polyline(n,N,x,-50,"Wind for T#sdbeat#t=0.5 years"); if(Type>1) Polyline(s,S);
F[6]=N[t+t];F[7]=N[t+t+1]; F[8]=N[m+m]; F[9]=N[m+m+1];
T=0.37*Year; Wind(); f=-Fmax/4; Findf(); t=m;f=Fmax/4; Findf();
SetPen(2,0x9933fa); Polyline(n,N,x,-70,"Wind for T#sdbeat#t=0.37 years"); if(Type>1) Polyline(s,S);
F[10]=N[t+t];F[11]=N[t+t+1]; F[12]=N[m+m];F[13]=N[m+m+1]; F[14]=90;F[15]=0;
SetPen(3,0xff0000); Polyline(8,F,x,-90,"Prediction");
TextHang(x,90,0,"The first ridge= % % d}\mp@subsup{}{}{\circ}\textrm{N}",k)
}
Wind() {n=0;s=0;
for(i=0;i<90;i+=1) { a=i*PI/180; v1=rot*rs*}\operatorname{cos}(\textrm{a}); v2=rot*rs
w=369.788-v2*}\operatorname{cos}(k*PI/180); a=v2*v2-4*PI*H/T; V=sqrt(a)-v1-w
if(V>-40 && V<60) {N[n+n]=i;N[n+n+1]=V; n+=1;}}
for(i=0;i<90;i+=1) { a=-i*PI/180;v1=rot*rs* cos(a); v2=rot*rs;
w=369.788-v2*\operatorname{cos(k*PI/180); a=v2*v2-4*PI*H/T; V=sqrt(a)-v1-w;}
if(V>-40 && V<60) {S[s+s]=-i; S[s+s+1]=V; s+=1;}}}
Findf(){a=1e10; for(i=0;i<n;i+=1) {b=N[i+i+1]-f;if(b<0) b=-b;if(b<a) {m=i;a=b;}}
}
#v07=?>A
```

For further improvement of precision, the value of the wind required by the constructive interference condition should be understood as a magnitude, it should be resolved into three components in the spherical coordinates $(r, A, \varphi)$ as

$$
\begin{equation*}
v_{\text {wind }}^{2}=v_{r}^{2}+v_{A}^{2}+v_{\varphi}^{2} \tag{29}
\end{equation*}
$$

According to the energy equipartition theorem in thermodynamics, approximately we have the average estimation

$$
\begin{equation*}
\left\langle v_{r}^{2}\right\rangle=\left\langle v_{A}^{2}\right\rangle=\left\langle v_{\varphi}^{2}\right\rangle=\frac{1}{3} v_{\text {wind }}^{2} . \tag{30}
\end{equation*}
$$

Thus, the wind vectors around the northern hemisphere of the Earth are plotted in Fig.11(b), the atmospheric circulation consists of three cells: Hadley cell, Ferrel cell, and arctic cell.


Fig. 12 NCEP/NCAR data, mean west winds over 40 years (1958~1997) [36]. (a) in winter; (b) in summer.

The beat $T_{\text {beat }}=0.5$ (years) blows comfortable winds over Europe, Northern America and Northeastern Asia, and modulates the four seasons; the shortest beat $T_{\text {beat }}=0.37$ (years) has a beat wavelength too long to be confined in the arctic regions so that it escapes from the north pole toward the equator, so recognized as the planet-scale waves or Rossby waves.

Since the acceleration-roll wave of the air interferes with the acceleration-roll wave of the equatorial shell, the easterlies at the equator have a magnitude of about $10 \mathrm{~m} / \mathrm{s}$. The trade winds or easterlies are the permanent east-to-west prevailing winds that flow in the Earth's equatorial region. The trade winds blow mainly from the northeast in the Northern Hemisphere and the southeast in the Southern Hemisphere, strengthening during the winter and when the Arctic oscillation is in its warm phase. Trade winds have been used by captains of sailing ships to cross the world's oceans for centuries. The driving force of atmospheric circulation is the uneven distribution of solar heating across the earth, which is greatest near the equator and least at the poles. This air rises to the tropopause, about $10-15$ kilometers above sea level, where the air is no longer buoyant [33].

Consider a funny issue, as we have known that $v_{\text {sun_effect }}=-84(\mathrm{~m} / \mathrm{s})$ at the equator when the first ridge at $A_{1}=12^{\circ} \mathrm{N}$, imagine if nuclear wars happen on the Earth to stop the solar radiation to the Earth's surface, then the equatorial wind on the Earth's surface will simply reach to 94 $(\mathrm{m} / \mathrm{s})$, like the winds on the surfaces of the Jupiter and Saturn [28], this will change global climate mode, killed all dinosaurs by strong winds and dusts. Thinking about Mars, Jupiter, and Saturn that gain very weak solar radiation, and there are at least $100(\mathrm{~m} / \mathrm{s})$ strong winds on their surfaces, as shown in Fig.10(b). It tells us how important is the easterly $10(\mathrm{~m} / \mathrm{s})$ at the equator for us, and how important is the air pollution for us. Have you ever gotten an experience: the more serious the air pollution, the stronger the wind? This section helps us understand how the dinosaurs to be buried.

## 8. Inner structure of tropical cyclones

A tropical cyclone is a rapidly rotating storm system characterized by a low-pressure center, a closed low-level atmospheric circulation, strong winds, and a spiral arrangement of thunderstorms that produce heavy rain and squalls. Tropical cyclones on either side of the Equator generally have their origins in several tropical cyclone basins, as shown in Fig.13. The Northwest Pacific Ocean is the most active basin on the planet, accounting for one-third of all tropical cyclone activity. Warm sea surface temperatures are required for tropical cyclones to form and strengthen. The commonly-accepted minimum temperature range for this to occur is


Fig. 13 Initial tropical cyclone events in the northern sphere and southern sphere.[35]

The latitude $A=12^{\circ} \mathrm{N}$ is called the first constructive interference ridge sandwiched between easterlies and westerlies, where the shear action of the winds will produce a lot of vortexes if the winds are disturbed by vapors at higher altitudes. The pregnancy of a tropical cyclone needs three steps, as shown in Fig.14(a).

Ready

(a)

Fig. 14 (a)Three steps of the pregnancy of a tropic cyclone in the northern hemisphere. (b) Dimension change on a molecular scale.

```
<Clet2020 Script>// [26]
double r,x,y,z,v, a,b,c,dP[20],D[500]; int i,j,k;
main() {'SetViewAngle("temp0, theta60,phi-40"); DrawFrame(FRAME_LINE,2,0xafff);
dP[0]=PEARL; dP[1]=0xefefef; dP[2]=1; dP[3]=XYZ;
dP[4]=10;dP[5]=10;dP[6]=0;dP[7]=50;dP[8]=30;dP[9]=30;
r=100;b=10;j=0;
for(i=0;i<360;i+=15) {a=i*PI/180; x=r*}\operatorname{cos}(a);y=r*\operatorname{sin}(a);z=0
x+=b*random(); y+=b*random();
D[j+j+j]=x;D[j+j+j+1]=y;D[j+j+j+2]=z; Plot(dP,D[j+j+j]); j+=1;}
Plot("POLYGON,1,@j,XYZ,10,",D);
r=80; b=6;dP[1]=0xff7f50; j=0;
for(i=0;i<360;i+=15) {a=i*PI/180; x=r*\operatorname{cos(a); y=r*sin(a); z=0;}
D[j+j+j]=x;D[j+j+j+1]=y; D[j+j+j+2]=z; Plot(dP,D[j+j+j]); j+=1;}
Plot("POLYGON,1,@j,XYZ,10,",D);
TextHang(10,0,70,"Ring2#n#n#nRing1");
TextAt(100,200,"Ring2 \longrightarrow Ring1,dimension chang from (V#sdr#t,V#sd\varphi#t) to (V#sd\varphi#t), must rotate,#nreleasing the latent
heat.");
D[0]=0; D[1]=0; D[2]=0; dP[1]=0x03a89e; dP[4]=30; dP[5]=30; Plot(dP,D);
TextHang(8,8,0,"Imagined #ndark mass"); r=70; j=0;
for(i=10;i<60;i+=5){a=i*PI/180; x=r* cos(a); y=r* sin(a); z=0;
D[j+j+j]=x;D[j+j+j+1]=y;D[j+j+j+2]=z; j+=1;}
dP[0]=POLYLINE; dP[1]=3; dP[2]=j; dP[3]=XYZ; dP[4]=15; Plot(dP,D);
```

Step1, during summer, the richer vapor at higher altitudes over the warm sea surface dramatically absorbs the solar radiation and releases the $v_{\text {sun_effect }}=-85(\mathrm{~m} / \mathrm{s})$ into the air at lower altitudes; consequently, the strengthened easterlies make a distortion to the first constructive interference ridge, as shown in Fig.14(a).

Step2, day by day, the distortion develops to an extent that it is going to separate from the mother-like first constructive interference ridge hit strongly by the westerlies, as shown in Fig.14(a).

Step3, finally, the distorted constructive interference ridge grows up to become an isolated baby ring which is recognized as a new tropical cyclone, counter-clockwise in the northern hemisphere, as shown in Fig.14(a), similar to Fig.10(b) there is a large vortex on the surface of Jupiter.

The pregnancy of tropical cyclones tends to develop at the latitude $A=12^{\circ} \mathrm{N}$, it is a relativistic quantum mechanical effect that cannot be solved by classical fluid mechanics. Recalling that the latitude $A=12^{\circ} \mathrm{N}$ represents the mean latitude angle which is subjected to the main solar radiation activity in the northern hemisphere, actually, the solar radiation varies its mostly shined latitudes within the range of $A=-23.5^{\circ} \mathrm{N} \sim 23.5^{\circ} \mathrm{N}$ due to the tilt of the earth axis concerning the earth orbital plan, therefore actual pregnancy latitude of tropical cyclones occur among latitude angles $A=-23.5^{\circ} \mathrm{N} \sim 23.5^{\circ} \mathrm{N}$, this result agrees well with the experimental records.

In Fig.11(a), there are the second, the third, ..., constructive interference ridges to bear the pregnancy of cyclones, hurricanes, typhoons, storms, tornados. Without these flexible constructive interference ridges, it is difficult for classical physics to say that some disturbances would somehow support the persistence of the vortexes.

For understanding the pregnancy process on a molecular scale, considering there are two molecular rings in Fig.14(b), typically the air molecules move on the Earth's surface in 2D motion. From ring2 to ring1, there is a dimension change for the molecules whose velocity changes from $\left(v_{r}, v_{\varphi}\right)$ to $\left(v_{\varphi}\right)$, from 2 D to 1 D , losing the $r$ component velocity. According to the energy equipartition theorem, every molecule would lose its kinetic energy as the latent heat released into the sky as

$$
\begin{equation*}
E_{\text {latent }}=\frac{2}{2} k T-\frac{1}{2} k T=\frac{1}{2} k T \tag{31}
\end{equation*}
$$

At this moment, the molecules have to rotate about the cyclone's center. Losing an amount of the molecular kinetic energy will lead to the ring1 to be in a stationary bound state, whose binding force and binding energy are given by

$$
\begin{align*}
& F_{\text {centripetal }}=-\frac{\Delta E}{\Delta r}=-\frac{N_{\text {molecules }} E_{\text {latent }}}{\Delta r}  \tag{32}\\
& E_{\text {binding }}=N_{\text {molecules }} E_{\text {latent }}
\end{align*}
$$

As you wish, you can say that at the cyclone's center there exists an imagined dark mass which contributes to this binding force by using the universal gravitational formula.

When the constructive interference ring of a newly born cyclone forms, its wavelength will adapt to the ring size as

$$
\begin{align*}
& \frac{1}{h M_{\text {cyclone }}} \oint_{L} v_{l} d l=2 \pi n ; \quad n=1,2, \ldots \\
& \psi=\exp \left(\frac{i}{h M_{\text {cyclone }}} \int_{0}^{x}\left(u_{1} d x_{1}+u_{2} d x_{2}+u_{3} d x_{3}+u_{4} d x_{4}\right)\right) \tag{33}
\end{align*}
$$

where the mass $M$ represents the overall mass of the new cyclone, including the imagined dark mass which accounts for the latent heat released during its formation; the constant $h$ is the Planck-constant-like constant determined by experimental observations. The air molecules of the ring are under the control of their acceleration-roll waves whose coherent length is so long that the waves have to overlap as

$$
\begin{align*}
& \psi(r)=1+e^{i \delta}+e^{i 2 \delta}+\ldots+e^{i(N-1) \delta}=\frac{1-\exp (i N \delta)}{1-\exp (i \delta)} \\
& \delta(r)=\frac{1}{h M_{\text {cyclone }}} \oint_{L}\left(v_{l}\right) d l \tag{3}
\end{align*}
$$

This formula can be used to calculate the distribution of the cyclone's nucleon density. In the preceding section on the central state optical model, we have mentioned that the air-vapor only concerns one wavelet interference, so that the overlapping number $N$ of air-vapor of the cyclones simply takes 2 , the structure of tropical cyclones becomes easier to calculate, that is

$$
\begin{equation*}
\psi(r)=1+e^{i \delta}=1+\cos (\delta)+i \sin (\delta) . \tag{35}
\end{equation*}
$$

It follows that the full expression is given in 2D cylinder wave form by

$$
\begin{equation*}
\psi(r)_{\text {interference }}=\sqrt{\frac{r_{0}}{r}} \sin (\delta) e^{i(k, r-\omega t)}=\sqrt{\frac{r_{0}}{r}} \sin \left(\frac{\nu 2 \pi r}{h M}\right) e^{i\left(k_{r} r-\omega t\right)} . \tag{3}
\end{equation*}
$$

On an altitude scale, a cyclone is divided into two layers: the upper layer and the lower layer. In the lower layer at the first ring the radius $r_{0}$ is proportional to the wavelength $\lambda$ of the acceleration-roll wave, the swirling wind speed approximately is inversely proportional to radius $r$, and the maximal speed is about $45(\mathrm{~m} / \mathrm{s})$ according to the theoretical prediction of the maximal wind in Fig.11(a) and the experimental observations in Fig.15(a), then


Fig, 15 (a)Gradient wind as a function of the radius for the hurricane DIANA (12, Sep.1984) measured at 850 hPa
[40]. (b) The simulation of a cyclone using acceleration-roll wave, $|x|<200 \mathrm{~km}$.

[^3]$21,118,20,121,19,124,20,127,21,130,21,133,20,136,20$,
main()\{SetAxis(X_AXIS, 0,0,200, "Radius\#n (km);0;50;100;150;200;");
SetAxis(Y_AXIS, $\overline{0}, 0,60, " W i n d ~(\mathrm{~m} / \mathrm{s}) ; 10 ; 20 ; 30 ; 40 ; 50 ; 60 ; ")$;
DrawFrame (FRAME BOX, 1,0xafffaf); $j=46$;
SetPen(2,0xff); Polyline(j,D Wind); SetPen(1,0xff0000);
$\mathrm{nP}[0]=\mathrm{CROSS} ; \mathrm{nP}[1]=0 ; \mathrm{nP}[\overline{2}]=\mathrm{j} ; \mathrm{nP}[3]=\mathrm{XY} ; \mathrm{nP}[4]=4 ; \mathrm{nP}[5]=4 ; \operatorname{Plot}\left(\mathrm{nP}, \mathrm{D} \_\right.$Wind $) ;$
TextHang(10,50,0,"DIANA, 12 Sep.1984, 850hPa.");
\}\#v07=?>A\#t
$<$ Clet2020 Script>//[26]
int i,j,k,nP[10]; double r_unit,Lamda,r,r0,r1,r2,v,v0,v1,w, a,b,delta, D[100];//1D array
main () $\left\{\mathrm{w}=150 ; \mathrm{r}_{\text {_ unit }}=1 \mathrm{e} 3 ; / / \mathrm{km}\right.$
$\mathrm{r} 0=30 * \mathrm{r}$ unit; $\mathrm{rl}=\mathrm{=} 0 ;$ Lamda $=2 * \mathrm{PI} * \mathrm{r} 0 ; \mathrm{v} 0=15 ; \mathrm{v} 1=30 ; / / \mathrm{m} / \mathrm{s}$
DrawFrame(FRAME_NULL,1,0xafffaf); $n P[0]=S E T ; n P[1]=1 ; n P[2]=P X$;
for $(\mathrm{i}=-\mathrm{w} ; \mathrm{i}<\mathrm{w} ; \mathrm{i}+=1)\left\{\operatorname{for}(\mathrm{j}=-\mathrm{w} ; \mathrm{j}<\mathrm{w} ; \mathrm{j}+=1)\left\{\mathrm{r} 2=\mathrm{i} * \mathrm{i}+\mathrm{j} * \mathrm{j} ; \mathrm{r}=\mathrm{sqrt}(\mathrm{r} 2){ }^{*} \mathrm{r}\right.\right.$ _unit;
if $(\mathrm{r}>\mathrm{r} 0) \quad\{\mathrm{b}=\mathrm{r} 0 / \mathrm{r} ; \mathrm{v}=\mathrm{v} 0+\mathrm{v} 1 * \mathrm{~b} ;\}$ else $\{\mathrm{b}=0 ; \mathrm{v}=\mathrm{v} 0 * \mathrm{~b} ;\}$
delta $=2 * \mathrm{PI} * \mathrm{v} * 2 * \mathrm{PI} * \mathrm{r} /(\mathrm{v} 0 *$ Lamda $)$;
$\mathrm{a}=\sin$ (delta); $\mathrm{a}^{*}=\mathrm{a} ; \mathrm{a}^{*}=\mathrm{b} ; / / 2 \mathrm{D}$ cylinder wave attenuation
nP[3]=Colorize(1,0xffffff,a);D[0]=250+i;D[1]=250+j;D[2]=0;PixelJob(nP,D); \}\}
Draw("LINE, $0,2, X Y, 10 ", " 0,0,100,0 ") ;$ TextHang(50,5,0,"200km");
\} $\# \mathrm{v} 07=$ ? $>\mathrm{A} \# \mathrm{t}$
\[

$$
\begin{align*}
& 2 \pi r_{0}=\lambda ; \quad v_{r}=\frac{h M}{2 \pi} k_{r}  \tag{37}\\
& r>r_{0}: \quad v_{\varphi}=v_{0}+v_{1} \frac{r_{0}}{r} ; \quad v_{0}=15(\mathrm{~m} / \mathrm{s}) ; \quad v_{1}=30(\mathrm{~m} / \mathrm{s})
\end{align*}
$$
\]

A cyclone's acceleration-roll wave interference is simulated as shown in Fig.15(b), clearly showing the inner structure of a cyclone, well compared with the DIANA cyclone on 12 Sep . 1984 in situ observation measured by an aircraft [40]. Don't underestimate the profound role of the Planck-constant-like constant in cyclone structure; without it, all return to the classical fluid mechanics.

What is dark matter? Why it is important for cyclones and galaxies? Consider a galaxy whose gravity can captures any stars near the galaxy center by the universal gravitational force; the near field of the gravity can capture any star strolling in the galaxy's 2D plane and finally let this star move in a 1D circular orbit approximately, it makes a dimension change for the star from 2D to 1 D with releasing latent heat. The far field of the gravity has to face stars strolling in 3D space, if it wants to capture these far stars, it must have an ability stronger than the universal gravitational force; in the far field, the gravitational force must become to be proportional to $1 / r$, which thus can capture far stars by releasing more latent heat in comparison to capturing near stars. Therefore, the gravitational force of the galaxy should be modified as

$$
f=a \frac{G m M^{\prime}}{r}+m \frac{G M}{r^{2}}=\left\{\begin{array}{cl}
m \frac{G M}{r^{2}} & \text { capturing 2D to1D }  \tag{38}\\
a \frac{G m M^{\prime}}{r} & \text { capturing 3D to1D }
\end{array}\right.
$$

where the coefficient $a$ only works in the far field of the gravity of the galaxy. In the viewpoint of the classical universal gravitational force, the far field attributes more unseen mass to the gravity effect, the unseen mass is recognized as the dark matter of the galaxy.

## 9. Sunspot formation

This section studies sunspot formation by comparing it with tropical cyclones on the earth.

These cyclones have established the following knowledge:
(1) tropical cyclone pregnancy basins are at a constructive interference ridge sandwiched between easterlies and westerlies, while zero wind in the ridge.


Fig. 16 (a)Sunspot basins: latitudes vs. time. (b) In northern hemisphere, the first sun's constructive interference ridge swings between $0^{\circ} \mathrm{N}$ and $30^{\circ} \mathrm{N}$, the second constructive interference ridge stays at the north pole.
<Clet2020 Script>// [26]
double beta,H,M,dP[20],D[2000],r,rs,rot,x,y,v1,v2,K1,K2,T1,T2,T,Lamda,V; int i,j,k,N;
int main() $\{$ beta $=2.961520 \mathrm{e} 10 ; \mathrm{H}=$ SPEEDC*SPEEDC*SPEEDC/beta;
$\mathrm{M}=1.9891 \mathrm{E} 30 ; \mathrm{rs}=6.95 \mathrm{e} 8 ; \mathrm{rot}=2 * \mathrm{PI} /(25.05 * 24 * 3600) ; \mathrm{vl}=\mathrm{rot} * \mathrm{rs} ; \mathrm{K} 1=\mathrm{v} 1 * \mathrm{v} 1 / 2 ; / / \mathrm{T} 1=2 * \mathrm{PI} * \mathrm{H} / \mathrm{K} 1$;
v2 $=0.7346 * \operatorname{sqrt}($ BOLTZMANN*5700/MP) $+0.2485 *$ sqrt(BOLTZMANN*5700/(MP+MP));
$\mathrm{K} 2=\mathrm{v} 2 * \mathrm{v} 2 / 2 ; \mathrm{T} 2=2 * \mathrm{PI} * \mathrm{H} /(\mathrm{K} 2-\mathrm{K} 1) ; \mathrm{T}=\mathrm{T} 2 / 24 * 3600 * 365.2422$;
Lamda $=2 * \mathrm{PI} * \mathrm{H} /(\mathrm{v} 2-\mathrm{v} 1) ; \mathrm{V}=$ Lamda/T2;
SetViewAngle("temp0, theta60, phi-60");
DrawFrame(FRAME_LINE, 1,0xafffaf); Overlook("2,1,60", D);
for $(\mathrm{i}=0 ; \mathrm{i}<180 ; \mathrm{i}+=15)\{\mathrm{k}=0 ; \mathrm{K} 1=0 ; \mathrm{K} 2=\mathrm{i} ;$ Grid ()$;\}$
for $(\mathrm{i}=0 ; \mathrm{i}<180 ; \mathrm{i}+=15)\{\mathrm{k}=1 ; \mathrm{K} 1=\mathrm{i} ; \mathrm{K} 2=0 ; \operatorname{Grid}() ;\}$
SetPen( $3,0 \times f f 0000$ ); $\mathrm{k}=1 ; \mathrm{K} 1=5 ; \mathrm{K} 2=0 ;$ Grid ();K1=80; K2=0; Grid();
TextHang(10,30,60,"second ridge"); TextHang(50,50,10,"first ridge");
Grid ()$\{\mathrm{N}=50 ; \mathrm{K} 1 *=\mathrm{PI} / 180 ; \mathrm{K} 2 *=\mathrm{PI} / 180 ; \quad \mathrm{r}=60$;
if $(\mathrm{k}=0)\{\mathrm{v} 1=2 * \mathrm{PI} / \mathrm{N} ; \mathrm{v} 2=0 ;\}$ else $\{\mathrm{v} 1=0 ; \mathrm{v} 2=2 * \mathrm{PI} / \mathrm{N} ;\}$
for $(\mathrm{j}=0 ; \mathrm{j}<=\mathrm{N} ; \mathrm{j}+=1)\{\mathrm{k}=\mathrm{j}+\mathrm{j}+\mathrm{j}$;
$\mathrm{D}[\mathrm{k}]=\mathrm{r} * \sin (\mathrm{~K} 1) * \cos (\mathrm{~K} 2) ; \mathrm{D}[\mathrm{k}+1]=\mathrm{r} * \sin (\mathrm{~K} 1) * \sin (\mathrm{~K} 2) ; \mathrm{D}[\mathrm{k}+2]=\mathrm{r} * \cos (\mathrm{~K} 1)$;
$\mathrm{K} 1+=\mathrm{v} 1 ; \mathrm{K} 2+=\mathrm{v} 2 ;\} \operatorname{Plot}(" P O L Y G O N, 0,50, \mathrm{XYZ}, 10 ", \mathrm{D}) ;$
\} $\# \mathrm{v} 07=$ ? $>\mathrm{A}$

The sunspot basins latitude vs. time as shown in Fig.16(a) tells us that sun's constructive interference ridge swings between $0^{\circ} \mathrm{N}$ and $30^{\circ} \mathrm{N}$ in the northern hemisphere, so do in the southern hemisphere. The first ridge's latitude angle $A_{l}$ in the northern hemisphere is written as

$$
\begin{equation*}
A_{1}=A_{0}\left[1+\sin \left(\frac{2 \pi t}{T_{\text {sunspot }}}+t_{0}\right)\right] ; \quad A_{0}=15^{\circ} \mathrm{N} ; \quad T_{\text {sunspot }}=11 \text { years } \tag{39}
\end{equation*}
$$

with zero wind in the ridge. Zero wind also must hold at the two poles for the sake of geometric shape, therefore the north pole must accommodate the second ridge to stay, its latitude angle $A_{2}$ is supposed to be

$$
\begin{equation*}
A_{2} \approx 90^{\circ} N-0 \cdot \sin \left(\frac{2 \pi t}{T_{\text {pole }}}+t_{0}\right) ; \quad T_{\text {pole }}=? \tag{40}
\end{equation*}
$$

We immediately deduce that the second ridge's period $T_{\text {pole }}$ equals to either $2 T_{\text {sunspot }}$ or $0.5 T_{\text {sunspot }}$, depending on the winds on the solar surface. The solar magnetic fields on the two poles flip every 22 years, which betrays a mystery: most probably to be $T_{\text {pole }}=2 T_{\text {sunspot }}$, regarding the second ridge as an undergoing-force to drive the solar magnetic fields on the two poles.
(2) Making use of the coherent width concept, and the spherical symmetry of shell's acceleration-roll wave.

Consider the interference between solar gas on the convictive zone and the dense core's shell. The gas $\psi_{\text {gas }}$ at the latitude angle $A$ and the shell's $\psi_{\text {shell }}$ at the same latitude are given by

$$
\begin{align*}
& \psi(r, A)=\psi_{\text {gas }}(r, A)+C \psi_{\text {shell }}(r, A)=\psi_{\text {gas }}(r, A)+C \psi_{\text {shell_equator }}(r) \\
& T_{\text {beat }} \simeq \frac{4 \pi c^{3}}{\beta\left(v_{\text {gas }}^{2}-v_{\text {shell_equator }}{ }^{2}\right)}  \tag{41}\\
& v_{\text {shell_equator }}=\omega r \\
& v_{\text {gas }}=\omega r \cos (A)+v_{\text {wind }}+v_{\text {convection_effect }}
\end{align*}
$$

where $C$ represents the coupling constant which relates to their distance and mass fractions within their coherent width, their interference leads to a beat phenomenon. The positive wind denotes the direction from west to east. The shell's $\psi_{\text {shell }}$ takes at the equator due to the spherical symmetry of shell's acceleration-roll wave. We have known that there is a beat $T_{\text {beat }}=11$ years in the first constructive interference ridge with zero wind, using the above period formula we obtain the convective influence:

$$
\begin{align*}
& v_{v_{\text {convection }- \text { effect }}}=6100-\omega r \cos \left(A_{1}\right) ; \quad(\text { units }: m / s) \\
& A_{1}=A_{0}\left[1+\sin \left(\frac{2 \pi t}{T_{\text {beat }}}+t_{0}\right)\right] ; \quad A_{0}=15^{\circ} N \tag{42}
\end{align*}
$$

In other words, the gas is subjected to the convective influence which enforces the beat oscillation in the ridge to run at the period $T_{\text {beat }}=11$ (years).

It is not easy to maintain the constructive interference condition for these waves. The wind required for maintaining the beat $T_{\text {beat }}=11$ (years) near the first ridge is calculated as shown in Fig.17(a) (blue line and red line), this wind will be destroyed by destructive interference at higher latitudes. But, the wind will arise up again at the second ridge where the waves satisfy the constructive interference condition: at $A=90^{\circ} \mathrm{N}$ location where the beat $T_{\text {beat pole }}=22$ (years).


Fig. 17 (a)Calculation of west winds required for the beat 11 years and 22 years in the northern hemisphere. (b)
The atmospheric circulation in the northern hemisphere.
<Clet2020 Script>// [26]
double beta,H,M,r,rc, rs,rot,v1,v2, Year,T,Lamda, V,a,b,w,N[500],F[100]; int i, j, k, t, m, n, f,Fmax,Type,x;
int main() $\{$ beta $=2.961520 \mathrm{e} 10 ; \mathrm{H}=$ SPEEDC*SPEEDC*SPEEDC/beta;
$\mathrm{M}=1.9891 \mathrm{E} 30 ; \mathrm{rs}=6.95 \mathrm{e} 8$; rot $=2 * \mathrm{PI} /(25.05 * 24 * 3600)$; Year $=24 * 3600 * 365.2422 ; \quad \mathrm{x}=10$;
SetAxis(X AXIS, $0,0,90$,"Latitude\#n( ${ }^{\circ} \mathrm{N}$ );0;30;60;90;");
SetAxis(Y_AXIS,-1000,-1000,1000,"West wind (m/s);-1000;-800;-600;-400;-200;0;200;400;600;800;1000;");
DrawFrame (0x016a,Type,0xafffaf);//Polyline(2,"-90,0,90,0");

```
Check(15,k); if(k>30) k=30; if(k<0) k=0;
T=22*Year; Wind(); f=0; Findf(); t=N[m+m]; T=11*Year; Wind(); f=0; Findf();
SetPen(2,0xff); Polyline(n,N,x,-400,"Wind for T#sdbeat#t=11 year");
F[0]=N[0];F[1]=N[1]; F[2]=N[m+m];F[3]=N[m+m+1]; t=(t+F[2])/2;//midpoint of two ridges
t=t-F[2]+m; Fmax=N[t+t+1]; //TextAt(100,20,"t=%d, Fmax=%f",t,Fmax);
f=Fmax;Findf(); F[4]=N[m+m];F[5]=N[m+m+1];
T=22*Year; Wind(); f=-Fmax/10; Findf(); t=m;f=Fmax/10; Findf();
SetPen(2,0x80ff00); Polyline(n,N,x,-600,"Wind for T#sdbeat#t=22 years");
F[6]=N[t+t];F[7]=N[t+t+1]; F[8]=N[m+m];F[9]=N[m+m+1];
F[10]=90;F[11]=0;
SetPen(3,0xff0000); Polyline(6,F,x,-800,"Prediction");
TextHang(x,900,0,"The first ridge = % % }\mp@subsup{}{}{\circ}\textrm{N}", k)
Wind() { n=0;
for(i=0;i<90;i+=1) {a=i*PI/180; v1=rot*rs*}\operatorname{cos(a); v2=rot*rs; w=6100-v2*}\operatorname{cos}(\textrm{k}*\textrm{PI}/180)
a=4*PI*H/T+v2*v2; V =sqrt(a)-v1-w;
if(V>-400 && V<700) {N[n+n]=i;N[n+n+1]=V;n+=1;}}}
Findf() {a=1e10; for(i=0;i<n;i+=1) {b=N[i+i+1]-f;if(b<0) b=-b;if(b<a) {m=i;a=b;}}
#vv07=?>A
```

The maximal wind appears most probably at the midpoint of the two ridges, about $580 \mathrm{~m} / \mathrm{s}$, linking all characteristic points in Fig.17(a) we obtain the predicted wind-curve over the northern hemisphere. The predicted second ridge has weak winds near the north pole, which agrees well with the judgment about the flipping of sun's geomagnetic field at the north pole.

For further improvement of precision, the value of the wind required by the constructive interference condition should be understood as a magnitude, it should be resolved into three components in the spherical coordinates $(r, A, \varphi)$ as

$$
\begin{equation*}
v_{\text {wind }}^{2}=v_{r}^{2}+v_{A}^{2}+v_{\varphi}^{2} \tag{43}
\end{equation*}
$$

According to the energy equipartition theorem in thermodynamics, approximately we have the average estimation

$$
\begin{equation*}
\left\langle v_{r}^{2}\right\rangle=\left\langle v_{A}^{2}\right\rangle=\left\langle v_{\varphi}^{2}\right\rangle=\frac{1}{3} v_{\text {wind }}^{2} . \tag{44}
\end{equation*}
$$

Thus, the wind vectors around the northern hemisphere of the sun are plotted in Fig.17(b), the atmospheric circulation consists of two cells: first cell, and second cell.
(3) Pregnancy mechanism of sunspots like tropic cyclones. The first constructive interference ridge sandwiched between easterlies and westerlies, where the shear action of the winds will produce a lot of vortexes if the winds are disturbed by the convective zone. The pregnancy of a sunspot needs three steps, as shown in Fig.18(a).

Step1, during active season, the stronger radiation output at higher altitudes over the warm surface and releases the $v_{\text {convection_effect }}$ into the gas at lower altitudes; consequently, the strengthened easterlies make a distortion to the first constructive interference ridge, as shown in Fig.18(a).

Step2, day by day, the distortion develops to an extent that it is going to separate from the mother-like first constructive interference ridge hit strongly by the westerlies, as shown in Fig.18(a).

Step3, finally, the distorted constructive interference ridge grows up to become an isolated baby ring which is recognized as a new sunspot, counter-clockwise in the northern hemisphere, as shown in Fig.18(a), similar to the famous large vortex on the surface of Jupiter.

## Ready


(a)

(b)

Fig. 18 (a)Three steps of the pregnancy of a sunspot in the northern hemisphere. (b) Inner structure of a sunspot.

When the constructive interference ring of a newly born cyclone forms, its wavelength will adapt to the ring size as

$$
\begin{align*}
& \frac{1}{h M_{\text {cyclone }}} \oint_{L} v_{l} d l=2 \pi n ; \quad n=1,2, \ldots  \tag{45}\\
& \psi=\exp \left(\frac{i}{h M_{\text {cyclone }}} \int_{0}^{x}\left(u_{1} d x_{1}+u_{2} d x_{2}+u_{3} d x_{3}+u_{4} d x_{4}\right)\right)
\end{align*}
$$

where the mass $M$ represents the overall mass of the new cyclone, including the imagined dark mass which accounts for the latent heat released during its formation; the constant $h$ is the Planck-constant-like constant determined by experimental observations. The air molecules of the ring are under the control of their acceleration-roll waves whose coherent length is so long that the waves have to overlap as

$$
\begin{align*}
& \psi(r)=1+e^{i \delta}+e^{i 2 \delta}+\ldots+e^{i(N-1) \delta}=\frac{1-\exp (i N \delta)}{1-\exp (i \delta)} \\
& \delta(r)=\frac{1}{h M_{\text {cyclone }}} \oint_{L}\left(v_{l}\right) d l \tag{46}
\end{align*}
$$

This formula can be used to calculate the distribution of the cyclone's nucleon density.

## 10. Geomagnetic field

In the solar system, planetary orbits are recognized as constructive interference ridges sandwiched by easterlies and westerlies with respect to the planets. As we have known, it is not easy to maintain the constructive interference condition for these acceleration-roll waves, so that winds required for maintaining these ridges are given by

$$
\begin{equation*}
\frac{\beta}{c^{3}} \oint_{L}\left(v_{w i n d}+\sqrt{\frac{G M}{r}}\right) d l=2 \pi n \quad \Rightarrow \quad v_{\text {wind }}=\frac{c^{3} n}{\beta r}-\sqrt{\frac{G M}{r}} \tag{47}
\end{equation*}
$$

The winds are calculated as shown in Fig.19(a) for those inner planets (Mercury, Venus, Earth,
and Mars). The maximal winds are determined simply by evenly dividing the intervals of these ridges. The earth system also has ridge's winds as shown in Fig.19(b).



Fig. 19 (a) Calculation of west winds in the solar system. (b) Calculation of west winds in the earth system.

```
<Clet2020 Script>// [26
double beta,H,M,r,rc, rs, rot,v1,v2, Year,T,Lamda,V,a,b,w,Fmax,N[2000],F[100]; int i, j, k, t, m, n, f, x;
int main() {beta=2.961520e10; H=SPEEDC*SPEEDC*SPEEDC/beta;
M=1.9891E30; rs=6.95e8; rot=2*PI/(25.05*24*3600); Year=24*3600*365.2422;
r=H*H/(GRAVITYC*M); V=H/r; Lamda=2*PI*r; x=3; w=6000;
SetAxis(X AXIS,0,0,10,"Radius#n (#rsr/r#sdl#re#t);0;2;4;6;8;10;");
SetAxis(Y-
DrawFrame(0x0156,1,0xafffaf);//Polyline(2,"-90,0,90,0");
Check(15,k); if(k>24) k=24; if(k<0) k=0; f=0;
SetPen(2,0xff); f=0; for(j=0;j<8;j+=1) {Wind(); Polyline(n,N);}
F[f+f]=j-0.25; F[f+f+1]=0;f+=1;
SetPen(2,0xff0000);Plot("POLYLINE,1,7,XY,8,0",F); Polyline(f-6,F[12],6.2,4500,"Prediction");
TextHang(3,-5000,0,"Mercury Earth");
TextHang(4,-4000,0,"Venus Mars");
Wind() {n=0;a=H*H/(GRAVITYC*M);
for(i=1;i<900;i+=1) {r=i*a/10; v1=sqrt(GRAVITYC*M/r); v2=j*H/r; V=v2-v1;
if(V>-W && V<w) {N[n+n]=sqrt(i/10);N[n+n+1]=V; n+=1;}
//ClipJob(APPEND,"i=%d v1=%f v2=%f V=%f",i,v1,v2,V);
} if(j<3) n=0;
b=j-0.25;r=b*b*a;v1=sqrt(GRAVITYC*M/r); v2=j*H/r; V=v2-v1; if(j==0) {b=0;V=0;}
if(V>w) V=w; F[f+f]=b;F[f+f+1]=V;f+=1;
b=j+0.25;r=b* b*a;v1=sqrt(GRAVITYC*M/r); v2=j*H/r; V=v2-v1;
if(V<-w) V=-w; F[f+f]=b;F[f+f+1]=V;f+=1;
}#v07=?>A
<Clet2020 Script>// [26]
double beta,H,M,r,rc, rs, rot,v1,v2, Year,T,Lamda,V,a,b,w,Fmax,N[500],F[100]; int i, j, k, n, f;
int main() {beta=1.377075e+14; H=SPEEDC*SPEEDC*SPEEDC/beta;
M=5.97237e24; rs=6.371e6; rot=2*PI/(24*3600); Year=24*3600*365.2422,
r=H*H/(GRAVITYC*M); V =H/r; Lamda=2*PI*r; w=1000;
SetAxis(X_AXIS,0,0,3,"Radius#n (#rsr/r#sd1#re#t);0;1;2;3;");
SetAxis(Y-AXIS,-w,-w,w,"West wind (m/s);-1000;-500;0;500;1000;");
DrawFrame(0x0134,1,0xafffaf);//Polyline(2,"-90,0,90,0");
Check(15,k); if(k>24) k=24; if(k<0) k=0; f=0;
SetPen(2,0xff); f=0; for(j=0;j<3;j+=1) {Wind(); Polyline(n,N);}
b}=\textrm{j}-0.25;\textrm{F}[\textrm{f}+\textrm{f}]=\textrm{b};\textrm{F}[\textrm{f}+\textrm{f}+1]=0;\textrm{f}+=1
SetPen(2,0xff0000);Polyline(f,F,1.8,600,"Prediction");
TextHang(1,-600,0,"quasi-satellite's perigee");
TextHang(2,-200,0,"Moon");
}
Wind() {n=0;a=H*H/(GRAVITYC*M);
for(i=0;i<180;i+=1) {r=i*a/20; v1=sqrt(GRAVITYC*M/r); v2=j*H/r; V=v2-v1;
if(V>-w && V<w) {N[n+n]=sqrt(i/20);N[n+n+1]=V; n+=1;}
//ClipJob(APPEND," i=%d v1=%f v2=%f V=%f",i,v1,v2,V);
}
b}=\textrm{j}-0.25;\textrm{r}=\textrm{b}*\textrm{b}*\textrm{a};\textrm{v}1=sqrt(GRAVITYC*M/r); v2=j*H/r; V=v2-v1; if(j==0) {b=0;V=0;}
if(V>w) V=w; F[f+f]=b;F[f+f+1]=V;f+=1;
b=j+0.25;r=b* b* ; v1=sqrt(GRAVITYC*M/r); v2=j*H/r; V=v2-v1;
if(V<-w)V=-w; F[f+f]=b;F[f+f+1]=V;f+=1;
}#v07=?>A
```

Although the solar plasma winds can continually feed the easterlies and westerlies that wrap the planets, the ridge's winds can be easily destroyed by themselves with lagged speeds un-adaptative to orbital speed:

$$
\begin{equation*}
v=\sqrt{\frac{G M}{r}} . \tag{48}
\end{equation*}
$$

So that the ridge's winds consist of very thin gas wrapping their planets, almost in vacuum.
But in the interior of the sun, the solar easterlies in Fig.19(a) (negative west winds) can drive the electrons in molten matter to flow, whose current would generate the solar geomagnetic field, as shown in Fig. 20.


Fig. 20 West winds in the interior of sun or earth, easterlies are negative west winds.
<Clet2020 Script>// [26]
double dP[20],D[2000],r,v1,v2,K1,K2; int i,j,k,N;
int main() $\{\mathrm{N}=50 ; \mathrm{r}=60 ; \mathrm{dP}[0]=\mathrm{POLYLINE} ; \mathrm{dP}[1]=0 ; \mathrm{dP}[2]=\mathrm{N}+1 ; \mathrm{dP}[3]=\mathrm{XYZ} ; \mathrm{dP}[4]=16$;
SetViewAngle("temp0, theta60,phi-40");
DrawFrame(FRAME LINE,1,0xafffaf);Overlook("2,1,60", D);
Draw("ARROW,0,2,XXYZ,16","80,0,0,80,30,0"); TextHang(70,0,-20,"west winds");
SetPen(1,0xed9121); for( $\mathrm{i}=0 ; \mathrm{i}<=180 ; \mathrm{i}+=5$ ) $\{\mathrm{k}=0 ; \mathrm{K} 1=0$; K2=i; Grid(); $\}$
$\operatorname{for}(\mathrm{i}=0 ; i<=180 ; i+=5)\{\mathrm{k}=1 ; \mathrm{K} 1=\mathrm{i} ; \mathrm{K} 2=0 ; \operatorname{Grid}() ;\}$
$/ / \operatorname{TextAt}(10,10, " \mathrm{v} 1=\% \mathrm{~d}, \mathrm{v} 2=\% \mathrm{~d}, \mathrm{~T}=\% .2 \mathrm{f} y, \lambda=\% \mathrm{e}, \mathrm{V}=\% \mathrm{~d} ", \mathrm{v} 1, \mathrm{v} 2, \mathrm{~T}$, Lamda,V);
SetPen(3,0xff0000); k=1; K1=70; K2=0;r=30; dP[1]=3; Grid();
$\mathrm{K} 1=90 ; \mathrm{K} 2=0 ; \mathrm{r}=30 ; \mathrm{dP}[1]=3$; Grid(); K1=110; K2 $=0 ; \mathrm{r}=30 ; \mathrm{dP}[1]=3$; $\operatorname{Grid}()$;
Grid() $\left\{\mathrm{K} 1^{*}=\mathrm{PI} / 180 ; \mathrm{K} 2 *=\mathrm{PI} / 180\right.$;
$\operatorname{if}(\mathrm{k}=0)\{\mathrm{v} 1=\mathrm{PI} / \mathrm{N} ; \mathrm{v} 2=0 ;\}$ else $\{\mathrm{v} 1=0 ; \mathrm{v} 2=\mathrm{PI} / \mathrm{N} ;\}$
for $(\mathrm{j}=0 ; \mathrm{j}<=\mathrm{N} ; \mathrm{j}+=1)\{\mathrm{k}=\mathrm{j}+\mathrm{j}+\mathrm{j}$;
$\mathrm{D}[\mathrm{k}]=\mathrm{r} * \sin (\mathrm{~K} 1) * \cos (\mathrm{~K} 2) ; \mathrm{D}[\mathrm{k}+1]=\mathrm{r} * \sin (\mathrm{~K} 1) * \sin (\mathrm{~K} 2) ; \mathrm{D}[\mathrm{k}+2]=\mathrm{r} * \cos (\mathrm{~K} 1) ;$
$\mathrm{K} 1+=\mathrm{v} 1 ; \mathrm{K} 2+=\mathrm{v} 2 ;\} \operatorname{Plot}(\mathrm{dP}, \mathrm{D})$;
\} $\# \mathrm{v} 07=$ ? $>\mathrm{A} \# \mathrm{t}$

The same thing happens in the earth, the earth's easterlies in Fig.19(b) (negative west winds) can drive the electrons in molten matter to flow, whose current will take the responsibility for generating the earth's geomagnetic field, as shown in Fig. 20.

It is said, the solar geomagnetic field flips its orientation every 22 years. The sun at a high temperature 5700 K , not only its molten core is conductive but also its atmosphere is conductive, both the electron-winds in the molten core and atmosphere can contribute to the solar geomagnetic field. Which one dominates the magnetic field? According to the preceding section, cyclones in two poles control the orientation of the solar geomagnetic field, and the cyclone's orientation flips every 22 years. Therefore, the conductive atmosphere dominates the solar geomagnetic field.

The earth's geomagnetic field is difficult to change its orientation, because the earth's atmosphere is non-conductive, the earth's magnetic field is mainly generated by the electronwind in the earth's molten matter, its orientation aligns with the ridge pattern in the solar system.

Although some geologists speculate the existence of earth's magnetic reversals, however there has not been a reversal in nearly 800 kiloyears. In earth's history, reversals are possible only when its ridge's wind was assaulted and destroyed by internal high temperatures or external strong plasma winds from exotic events. Measurements indicate that earth's magnetic field has weakened by about $10 \%$ since 1800 s.

Now let us calculate the earth's magnetic field $B$ at the north pole. The earth radius is $r_{s}$, with angular speed $\omega=2 \pi /(24 * 3600)$; the radius of molten core takes $R=r_{s} / 2, n_{e}$ denotes the conducting electron density, electron-wind at latitude $A$ in the molten core is given by

$$
\begin{equation*}
v_{\text {wind }}=0-\omega r \cos A . \tag{49}
\end{equation*}
$$

According to the magnetic field formula of a single coil, as shown in Fig.21, the $B$ is given by


Fig. 21 The magnetic field of a single coil.
$<$ Clet2020 Script>// [26]
double beta,H,M,r,rc,rs,rot, ne,V,R,A,B,a,b,d,m,v1,v2,T,Year; int i, j,k;
int main() $\{$ Year $=24 * 3600 * 365.2422$;
beta $=2.961520 \mathrm{e} 10 ; \mathrm{H}=$ SPEEDC*SPEEDC*SPEEDC/beta;
$\mathrm{M}=1.9891 \mathrm{E} 30 ;$ rs $=6.95 \mathrm{e} 8$; rot $=2$ *PI/(25.05*24*3600);
beta=1.377075e+14; H=SPEEDC*SPEEDC*SPEEDC/beta;
$\mathrm{M}=5.97237 \mathrm{e} 24 ;$ rs=6.371 6 ; rot $=2$ *PI/(24*3600);
$\mathrm{B}=0 ; \mathrm{a}=\mathrm{rs} / 200 ; \mathrm{b}=\mathrm{PI} / 100 ;$ ne $=2.5 \mathrm{e} 13$
$\operatorname{for}(\mathrm{i}=1 ; \mathrm{i}<100 ; \mathrm{i}+=1)\{\mathrm{r}=\mathrm{a} * \mathrm{i} ;$ for $(\mathrm{j}=0 ; \mathrm{j}<100 ; \mathrm{j}+=1)\{\mathrm{A}=\mathrm{b} *(\mathrm{j}-50) ; \mathrm{R}=\mathrm{r} * \cos (\mathrm{~A}) ; \mathrm{V}=\mathrm{rot} * \mathrm{R}$;
$d=r s-r * \sin (A) ; d=R * R+d * d ; d=s q r t(d) ; d=2 * d * d * d ;$
$\mathrm{B}+=\mathrm{MC} * \mathrm{R} * \mathrm{R}$ *CHARGEC*ne*V*r/d; \}\}
$\mathrm{B}=\mathrm{B} * \mathrm{a} * \mathrm{~b} ; \mathrm{m}=\mathrm{ne} / \mathrm{AVOGADRO}$;
$\operatorname{TextAt}(100,100, " n e=\% e, \quad \mathrm{~m}=\% \mathrm{e}, \mathrm{B}=\% \mathrm{e}$ ", ne, $m, B)$;
$\mathrm{v} 1=\mathrm{rot} * \mathrm{rs} / 2 ; \mathrm{v} 2=0 ; \mathrm{T}=4 * \mathrm{PI} * \mathrm{H} /(\mathrm{v} 1 * \mathrm{v} 1-\mathrm{v} 2 * \mathrm{v} 2) ; \mathrm{T}=\mathrm{T} / \mathrm{Year}$;
TextAt(100,200,"v1=\%e, v2=\%e, T=\%f y,",v1,v2,T);
\} $\# \mathrm{v} 07=$ ? $>\mathrm{A} \# \mathrm{t}$

$$
\begin{align*}
& B=\int_{r=0}^{r=r_{s} / 2} \int_{A=-\pi / 2}^{A=\pi / 2} \frac{\mu_{0} R^{2} e n_{e} \nu_{\text {wind }} r d r d A}{2\left(R^{2}+\left(r_{s}-r \sin A\right)^{2}\right)^{3 / 2}} ; \quad R=r \cos A  \tag{50}\\
& n_{e}=2.5 e+13\left(m^{-3}\right) ; \quad B=0.6(\text { Gauss })
\end{align*}
$$

By fitting data, if the conducting electron density takes $n_{e}=2.5 \mathrm{e}+13\left(\mathrm{~m}^{-3}\right)$, then we obtain the earth's geomagnetic field at the north pole $B=0.6 \mathrm{Gauss}$, which is consistent with experimental observation. To note that the conducting electron density is very thin, about $4.15 \mathrm{e}-$ $11(/ \mathrm{mole})$. Mars and Venus are either too small to have a convective core or rotate very slowly, so they are with weak or no magnetic fields.

Smart hunters can quickly find rabbits, deer or foxes by tracking animals in the snow. As we know, wind is required to meet the conditions of constructive interference. Therefore, the appearance of a gust is the signal of quantum system working in the atmosphere. Extracting energy from wind and rivers actually sabotages the quantum stability of the Earth's atmosphere, which may inevitably change the global climate, as shown in Fig.22.


Fig. 22 Extracting energies from winds and rives actually sabotages the quantum stability of the earth's atmosphere, may inevitably change the global climate.

## 11. Conclusions

In analogy with the ultimate speed $c$, there is an ultimate acceleration $\beta$, nobody's acceleration can exceed this limit $\beta$, in the solar system, $\beta=2.961520 \mathrm{e}+10\left(\mathrm{~m} / \mathrm{s}^{2}\right)$. Because this ultimate acceleration is large, any effect related to $\beta$ will become easy to test, including quantum gravity tests. In this paper, an approach is put forward to connect the ultimate acceleration with quantum theory, and is applied to sunspot, earth's tropic cyclones, and earth's geomagnetic field problems. The sunspot cycle is calculated to be 10.93 years due to the ultimate acceleration. A simulation was carried out, clearly showing the inner structure of a cyclone, which is consistent with the famous DIANA cyclone on 12 September 1984 in situ observation measured by an aircraft. The sunspot structure is also investigated, which has a similarity to that of a tropic cyclone. The ultimate acceleration provides a mechanism to explain geomagnetic field, the earth's geomagnetic field is calculated to be 0.6 Gauss at the north pole.

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$6 \mathrm{rgcoMtOu} 9 \mathrm{w} 4 \mathrm{rP} 49 \mathrm{sgN} /$ view? $\mathrm{usp}=$ sharing
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[^0]:    $<$ Clet2020 Script $>/ /[26]$
    int $\mathrm{i}, \mathrm{j}, \mathrm{k}, \mathrm{m}, \mathrm{n}, \mathrm{N}, \mathrm{nP}[10]$;
    double beta,H,B,M,r,r_unit,x,y,z, delta,D[1000],S[1000], a,b,rs,rc,rot, atm height; char str[100];
    main() $\{\mathrm{k}=150 ; \mathrm{rs}=6.95 \mathrm{e} 8 ; \mathrm{rc}=0 ; \mathrm{x}=25.05 ; \mathrm{rot}=2 * \mathrm{PI} /(\mathrm{x} * 24 * 3600) ; \mathrm{n}=0 ; \mathrm{N}=3 \overline{4} 1$;
    beta $=2.961520 \mathrm{e} 10 ; \mathrm{H}=\mathrm{SPEEDC} * S P E E D C * S P E E D C / b e t a ; M=1.9891 E 30 ;$ atm_height=2e6; r_unit=1E7; $\mathrm{b}=\mathrm{PI} /\left(2 * \mathrm{PI}\right.$ * rot $^{*}$ rs*rs/H);
    for $(\mathrm{i}=-\mathrm{k} ; \mathrm{i}<\mathrm{k} ; \mathrm{i}+=1)\left\{\mathrm{r}=\mathrm{abs}(\mathrm{i})^{*} \mathrm{r}\right.$ unit;
    if( $\mathrm{r}<\mathrm{rs}+$ atm_height ) delta $=2 * \overline{\mathrm{PI}} * \mathrm{rrt}^{*} \mathrm{r}^{*} \mathrm{r} / \mathrm{H}$; else delta $=2 * \mathrm{PI}^{*} \operatorname{sqrt}(\mathrm{GRAVITYC} * \mathrm{M} * \mathrm{r}) / \mathrm{H} ; / /$ around the star
    $y=S u m J o b\left(" S L I T \_A D D, @ N, @ d e l t a ", D\right) ; y=y /(N * N) ;$
    $\mathrm{S}[\mathrm{n}]=\mathrm{i} ; \mathrm{S}[\mathrm{n}+1]=\mathrm{y} ; \mathrm{if}(\mathrm{i}>0 \& \& \mathrm{rc}==0 \& \& \mathrm{y}<0.001) \mathrm{rc}=\mathrm{r} ; \mathrm{D}[\mathrm{n}]=\mathrm{i} ; \mathrm{D}[\mathrm{n}+1]=\mathrm{z} ; \mathrm{n}+=2 ;\}$
    SetAxis(X_AXIS,-k,0,k,"\#ifr; ; ; ;");SetAxis(Y_AXIS,0,0,1.2,"\#if| $|\#| \# s u 2 \# t ; 0 ; 0.4 ; 0.8 ; 1.2 ; ")$;
    DrawFrame $\bar{e}(F R A M E$ SCALE, 1,0xafffaf); $z=1 \overline{0} 0 *(r s-r c) / r s$;
    $\operatorname{SetPen}(1,0 x f f 0000) ; \operatorname{Polyline}(\mathrm{k}+\mathrm{k}, \mathrm{S}, \mathrm{k} / 2,1$," nucleon_density" $) ; \operatorname{SetPen}(1,0 \mathrm{x} 0000 \mathrm{ff}) ; / / \operatorname{Polyline}(\mathrm{k}+\mathrm{k}, \mathrm{D})$;
    //Draw("LINE,0,2,XY,0","20,0.5,60,0.6");TextHang(60,0.6,0,"core");
    $\mathrm{r}=\mathrm{rs} / \mathrm{r}$ _unit; $\mathrm{y}=-0.05 ; \mathrm{D}[0]=-\mathrm{r} ; \mathrm{D}[1]=\mathrm{y} ; \mathrm{D}[2]=\mathrm{r} ; \mathrm{D}[3]=\mathrm{y} ;$ Draw("ARROW,3,2,XY,10,100,10,10,",D);
    Format(str,"\#ifN\#t $=\% \mathrm{~d} \# \mathrm{n} \#$ if $\beta \# t=\% \mathrm{e} \# \mathrm{nrc}=\% \mathrm{e} \# \mathrm{nrs}=\%$ e\#nerror $=\% .2 \mathrm{f} \% ", \mathrm{~N}$, beta, rc, rs, z );
    TextHang(k/2,0.7,0,str);TextHang(r+5,y/2,0,"\#ifr\#sds\#t");TextHang(-r,y+y,0,"Sun diameter");
    \}\#v07=? $>\mathrm{A} \# \mathrm{t}$

[^1]:    $<$ Clet2020 Script $>/ /[26]$
    int i,j,k,m,n,N,nP[10]; double H,B,M,v_r,r,AU,r_unit,x,y,z,delta,D[10],S[1000];
    double rs,rc,rot,a,b,atm height,beta; char str[100];
    main ()$\{\mathrm{k}=80 ; \mathrm{rs}=6.378 \overline{\mathrm{e}} ; \mathrm{rc}=0 ; \mathrm{atm}$ height $=1.5 \mathrm{e} 5 ; \mathrm{n}=0 ; \mathrm{N}=65$;
    beta $=1.377075 \mathrm{e}+14 ; \mathrm{H}=$ SPEEDC $*$ SPEEDC*SPEEDC/beta;
    $\mathrm{M}=5.97237 \mathrm{e} 24 ; \mathrm{AU}=1.496 \mathrm{E} 11 ; \mathrm{r}_{-}$unit=1e-6*AU; rot=2*PI/(24*60*60);//angular speed of the Earth
    for $(\mathrm{i}=-\mathrm{k} ; \mathrm{i}<\mathrm{k} ; \mathrm{i}+=1)\{\mathrm{r}=\mathrm{abs}(\mathrm{i}) * \mathrm{r}$ unit;
    
    delta=2*PI*v_r/H; y=-SumJob("SLIT_ADD,@N,@delta",D); y=y/(N*N);
    if $(\mathrm{y}>1) \mathrm{y}=1 ; \mathrm{S}[\mathrm{n}]=\mathrm{i} ; \mathrm{S}[\mathrm{n}+1]=\mathrm{y} ; \mathrm{if}(\mathrm{i}>0 \overline{\&} \& \mathrm{rc}==0$ \&\& $\mathrm{y}<0.001) \mathrm{rc}=\mathrm{r} ; \quad \mathrm{n}+=2 ;\}$
    SetAxis(X_AXIS,-k,0,k,"r; ;;;");SetAxis(Y_AXIS, 0,0,1.2,"\#ifl||\#su2\#t;0;0.4;0.8;1.2;");
    DrawFrame(FRAME SCALE, $1,0 x a f f f a f) ; x=50 ; z=100 *(r s-r c) / r s ;$
    SetPen(1,0xff0000);Polyline(k+k,S,k/2,1," nucleon_density");
    $\mathrm{r}=\mathrm{rs} / \mathrm{r}$ unit $; \mathrm{y}=-0.05 ; \mathrm{D}[0]=-\mathrm{r} ; \mathrm{D}[1]=\mathrm{y} ; \mathrm{D}[2]=\mathrm{r} ; \mathrm{D}[3]=\mathrm{y}$;
    SetPen(2,0x0000ff); Draw("ARROW,3,2,XY,10,100, 10, 10,",D);
     TextHang(k/2,0.7,0,str);TextHang(r+5,y/2,0,"r\#sds\#t");TextHang(-r,y+y,0,"Earth diameter"); \} $\# \mathrm{v} 07=$ ? $>\mathrm{A} \# \mathrm{t}$

[^2]:    $<$ Clet2020 Script>// [26]
    int i, j, k, R,D[500];
    main() \{DrawFrame(FRAME_NULL, 1,0xafffaf);
    $\mathrm{R}=60$; $\operatorname{SetPen}(1,0 x f f f f 00)$;
    D[0]=-R; D[1]=-R; D[2]=R; D[3]=R; Draw("ELLIPSE, 1,2,XY,0",D);
    $\mathrm{R}=85 ; \quad \mathrm{k}=15 ; \operatorname{Set} \operatorname{Pen}(1,0 \mathrm{xff0000})$;
    D $[0]=-\mathrm{R} ; \mathrm{D}[1]=-\mathrm{R} ; \mathrm{D}[2]=\mathrm{R} ; \mathrm{D}[3]=\mathrm{R} ;$ Draw("ELLIPSE, $0,2, \mathrm{XY}, 0 ", \mathrm{D})$;
    $\mathrm{D}[0]=0 ; \mathrm{D}[1]=0 ; \mathrm{D}[2]=\mathrm{R}-\mathrm{k} ; \mathrm{D}[3]=0 ; \mathrm{D}[4]=\mathrm{R}+\mathrm{k} ; \mathrm{D}[5]=0$;
    Draw("SECTOR, 1,3,XY, 15,30, 130,0",D);
    R=95; k=15; SetPen(1,0x00ff);
    D $[0]=-R ; D[1]=-R ; D[2]=R ; D[3]=R ;$ Draw("ELLIPSE, $0,2, X Y, 0 ", D) ;$
    $\mathrm{D}[0]=0 ; \mathrm{D}[1]=0 ; \mathrm{D}[2]=\mathrm{R}-\mathrm{k} ; \mathrm{D}[3]=0 ; \mathrm{D}[4]=\mathrm{R}+\mathrm{k} ; \mathrm{D}[5]=0$;
    Draw("SECTOR, 1,3,XY, 15,0,100,0",D);D[4]=R+k+k;
    Draw("SECTOR,3,3,XY, 15,0,100,0",D);
    TextHang( $0,0,0$,"dense core"); TextHang(R-k-k,-k, 0 ,"Coherent width");
    TextHang $(0, \mathrm{R}+\mathrm{k}+\mathrm{k}+\mathrm{k}, 0$, "Coherent length");TextHang(-R,R+k,, ,"Overlapping");
    \} $\# \mathrm{v} 07=$ ? $>\mathrm{A} \# \mathrm{t}$

[^3]:    <Clet2020 Script>//[26]
    int i,j,k,n; double x,y,nP[10];
    int
    D_Wind[92] $=\{2,5,4,15,7,23,10,32,13,39,16,44,19,44,22,40,25,37,28,34,31,32,34,32,37,32,40,32,43,32,46,32,49,32,52,31,55,30$, $5 \overline{8}, 28,61,27,64,26,67,26,70,26,73,25,76,24,79,24,82,25,85,25,88,24,91,24,94,22,97,21,100,21,103,20,106,20,109,21,112,21,115$,

